

Committee on Earth Observation Satellites
Working Group on Calibration and Validation

Microwave Sensors Subgroup

9 November 2005



Christopher Buck

- Payload Engineer
- Working at ESA/ESTEC since 12/87
- Main area of competence: Microwave Instruments for Earth Observation
- Succeeded Manuel Martín-Neira as MW Sensors chair





- **CryoSat Status**
- **SMOS Status**
- **Sentinels 1 & 3**
- **P-band Ice Sounder Demonstrator**
- **PARIS Airborne Demonstrator**
- **ASAR Receiver Gain Droop Experiment**
- **SurfSat**



Launched 8 October 2005

- Rockot from Plesetsk Cosmodrome

Failed during separation of 2nd and 3rd stages

Failure attributed to software error

- Reproduced on-ground
- Go-ahead for future launches given

Motivation for CryoSat 2 is high

- Programme will take 2-3 years

Experimental Campaigns with ASIRAS set to continue





Soil
Moisture
Ocean
Salinity



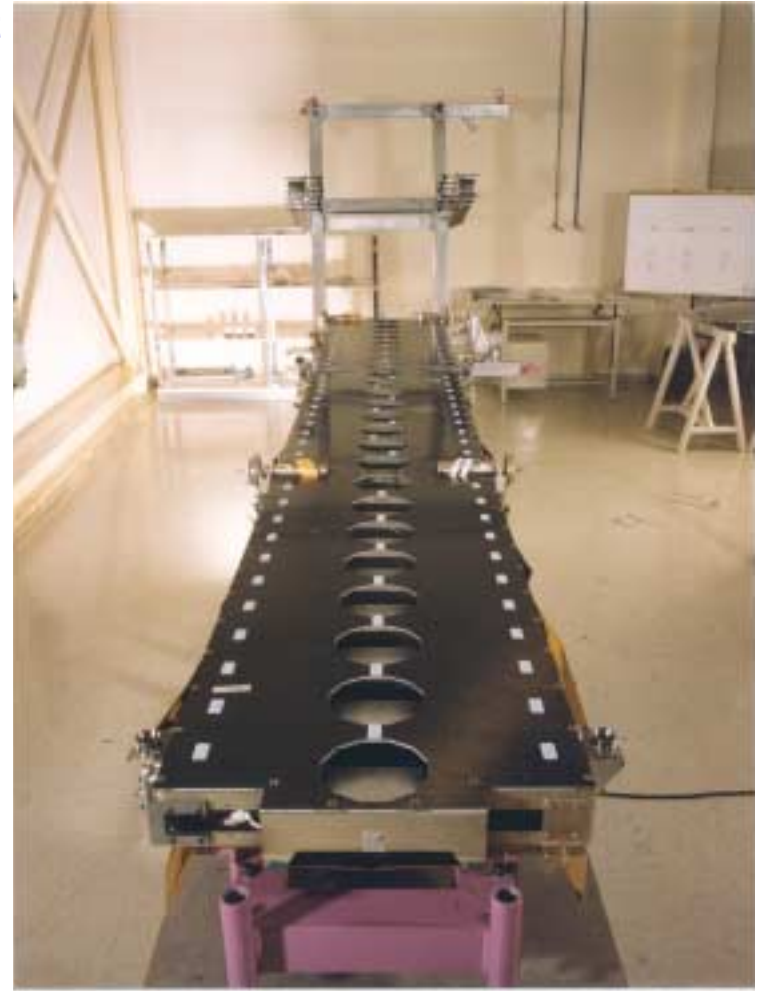


- 2nd mission within ESA's "Opportunity Missions in Earth Observation" programme
- Mission objectives to retrieve maps of:
 - **Soil Moisture**
 - **Ocean Salinity**
- SMOS is a collaboration between CDTI (Spain), CNES (France) and ESA
- Main mission elements:
 - **Platform = PROTEUS (CNES)**
 - **PLM, developed by ESA, EADS CASA Espacio as main contractor**
 - **VILLAFRANCA (Madrid), PLM operational centre**
 - **Eurockot, launcher**



Deployment mechanism test (October - November 2002)

Flight hardware LICEF assemblies

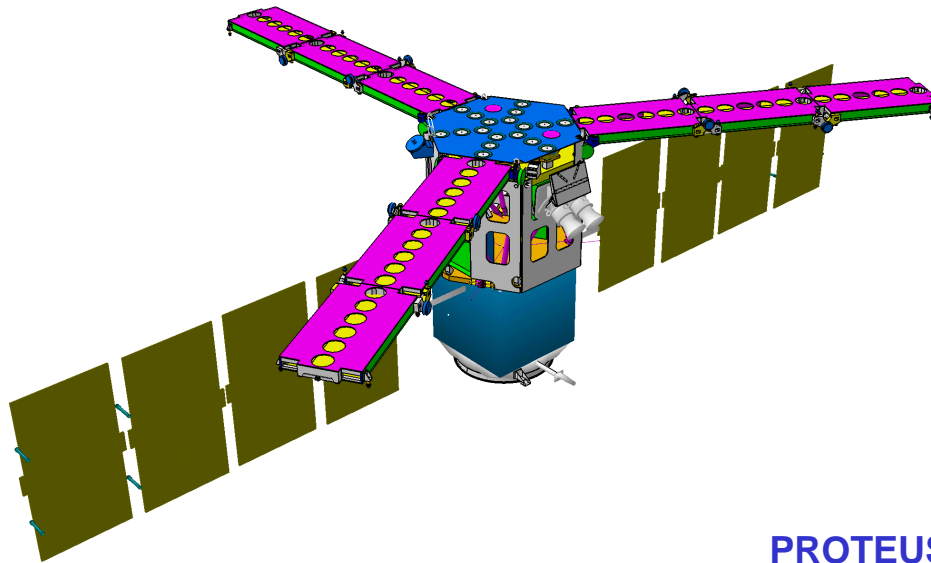




SMOS PLM

MASS: 360kg

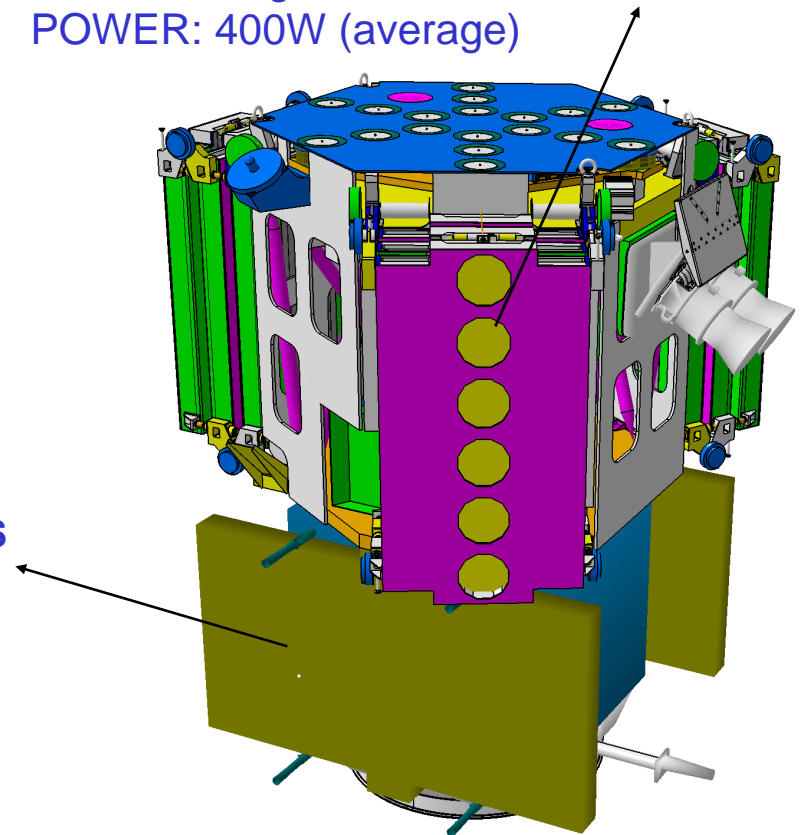
POWER: 400W (average)



SMOS = L-band RADIOMETER
(MIRAS)

PROTEUS

PLM



MIRAS = Microwave Imaging Radiometer with Aperture Synthesis



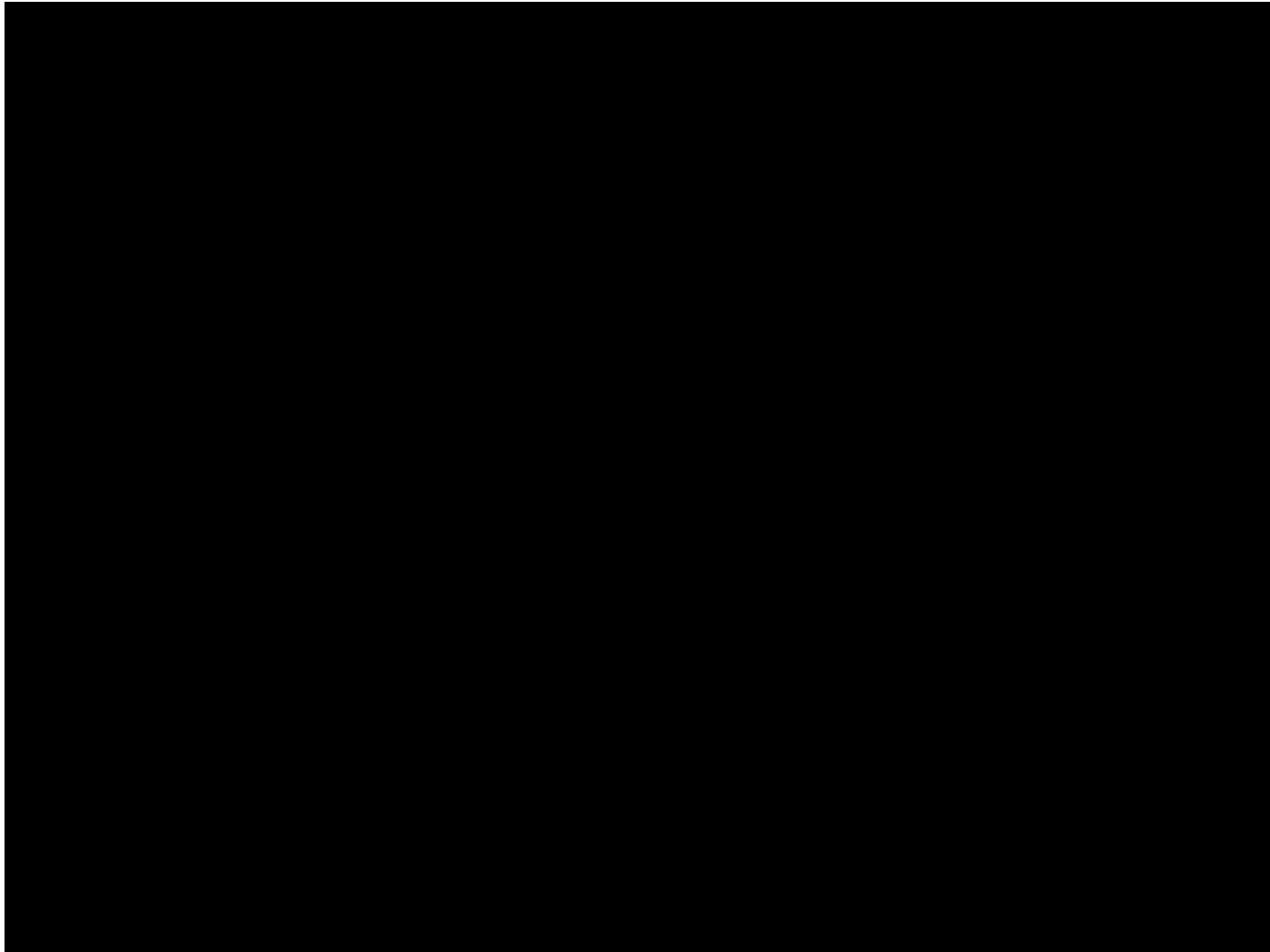
MIRAS PAYLOAD:

- MIRAS STM is at ESTEC for environmental tests
- MIRAS EM is undergoing end-to-end electrical tests at INTA
- MIRAS PAYLOAD CDR held in October 2005

PROTEUS PLATFORM

- PROTEUS Platform - MIRAS Payload Interface Verification
- Platform Flight Model will be integrated in September/December 2005
- SMOS Satellite Flight Model (Platform + PLM) will be integrated in November 2006 / July 2007

Launch – September 2007





- **Sentinel-1 “Red”**
 - C-band SAR, 100MHz BW, Quad-Pol
 - Trade-off on platform
 - PRIMA or Snapdragon
 - Phase A KO April 2005
 - Prime: EADS Astrium

- **Sentinel-2 “Blue”**
 - Ocean mission
 - Radar Altimeter + Imaging Spectrometer + Radiometer
 - Trade-offs on:
 - No of satellites
 - Type of altimeter (SIRAL, SRAL, RA-2)
 - No of radiometer channels (up to 3)
 - KO September 2005
 - Prime: Alcatel-Alenia Space



P Sounder Demonstrator (TRP/EOEP)

- Development of a P-band (435MHz) radar for ice sounding
 - Capable of penetrating up to 4km of ice
 - Radar will be installed in a Twin Otter aircraft
- 
- Proof-of-concept flight to be conducted over Greenland
 - Contract won by Oersted.DTU and DNSC but start delayed until 2/1/06



De-Havilland Twin Otter aircraft



□ Bi-static Radar:

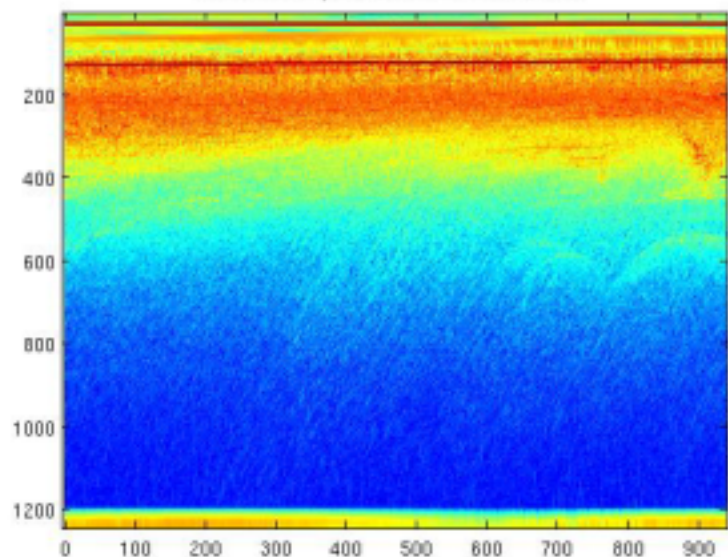
- 4 element transmit antenna array on port side
- Identical receive array on starboard side



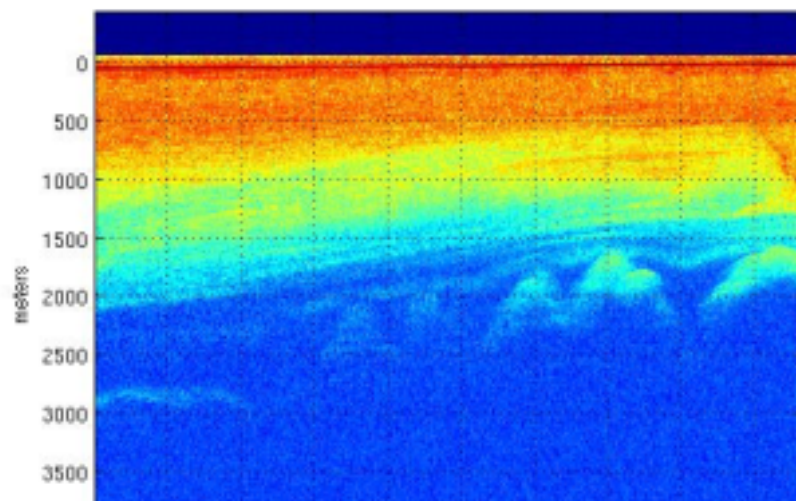
Description of Radar parameter	Characteristics/Value
RF carrier frequency	150 MHz
Chirp Bandwidth	10 MHz – 15 MHz
Transmitted pulse Width	Selectable (250ns-4 μ s)
PRF	Selectable (kHz)
Peak Transmit Power	1 kW into each of the four antennas
Antenna (bi- static)	8 folded dipole elements orientated across track : - 4 emitters - 4 receivers
Polarimetric mode	2 antennas on each side orientated along track



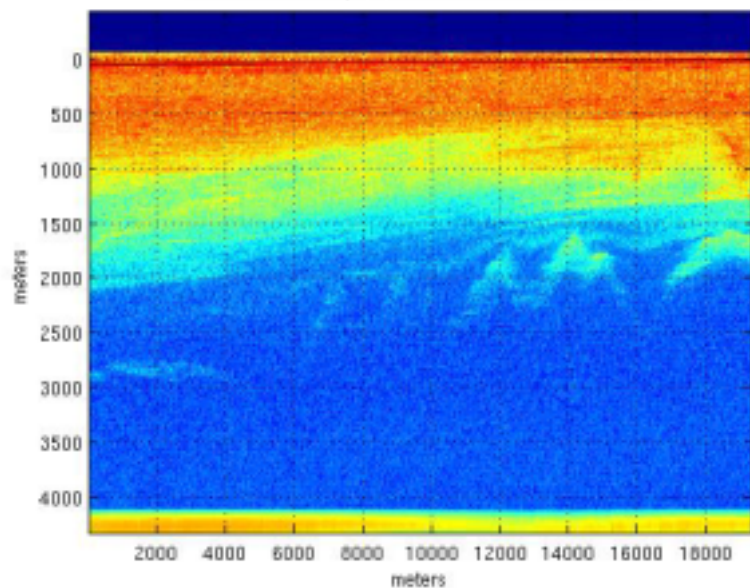
Raw Data Chirp Lower Channel files 158-173



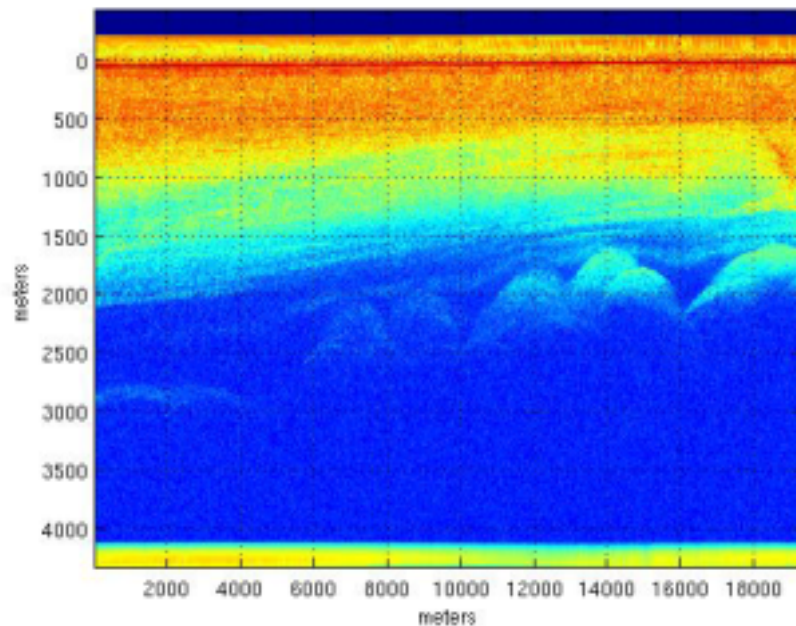
Focused SAR Chirp lower channel 158-173 - CL=100m

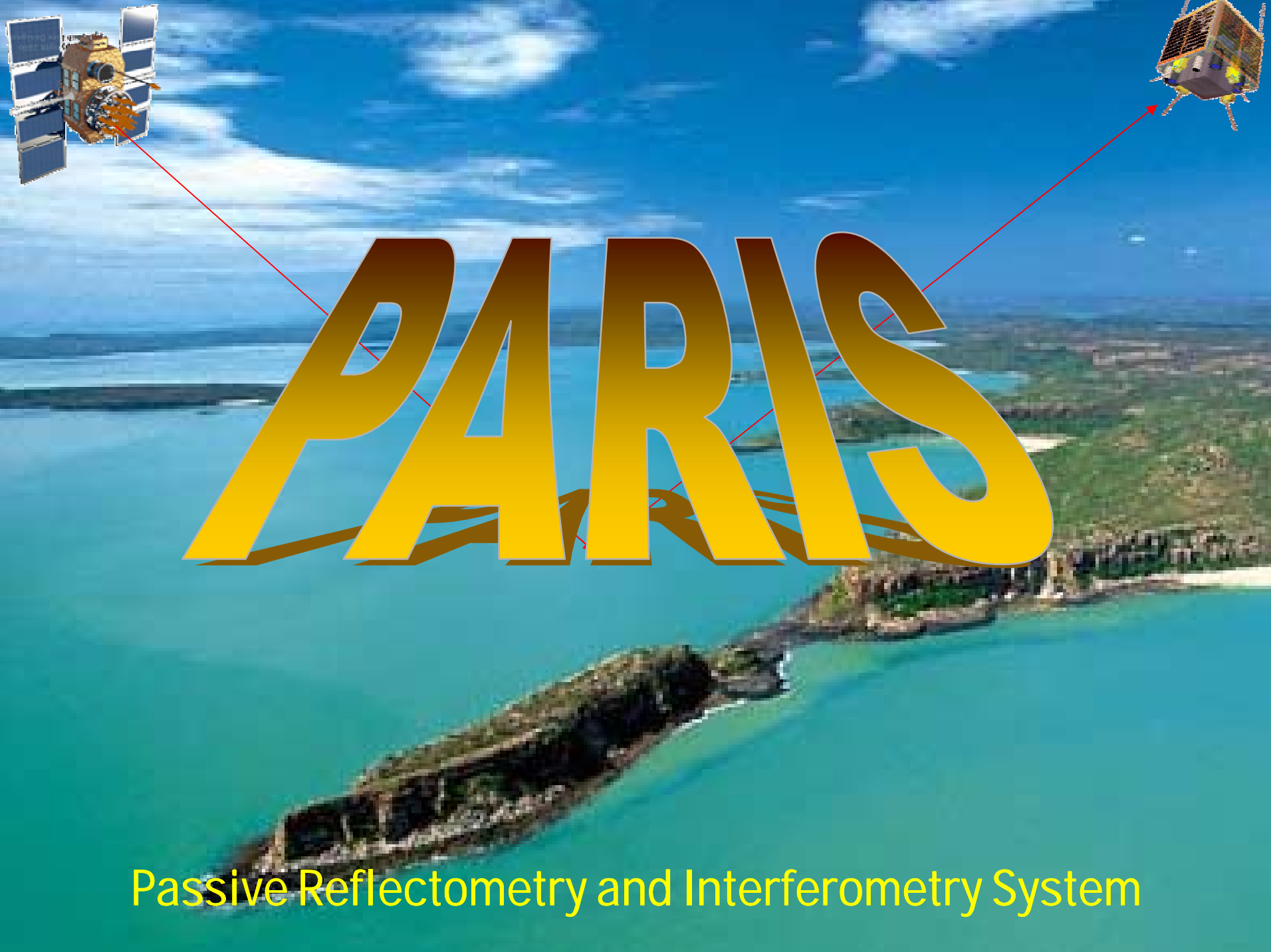


Focused SAR Chirp lower channel 158-173 - CL=50m



Focused SAR Chirp lower channel 158-173





PARIS

Passive Reflectometry and Interferometry System




- **PARIS Airborne Demonstrator (TRP – 1.75M)**
 - Development of a four-beam PARIS instrument for performing ocean altimetry using reflected GPS signals
 - Split into four main parts:
 - A – The overall project including Instrument Control Unit (Starlab) mounting into the aircraft (Do228) and flight – Astrium Portsmouth
 - B – Mutli-beam array – Q-Par Angus
 - C – Signal Processor – Austrian Aerospace, IEEC, D+T CNM
 - D – Digital Beamforming Network and Receiver – Astrium Stevenage



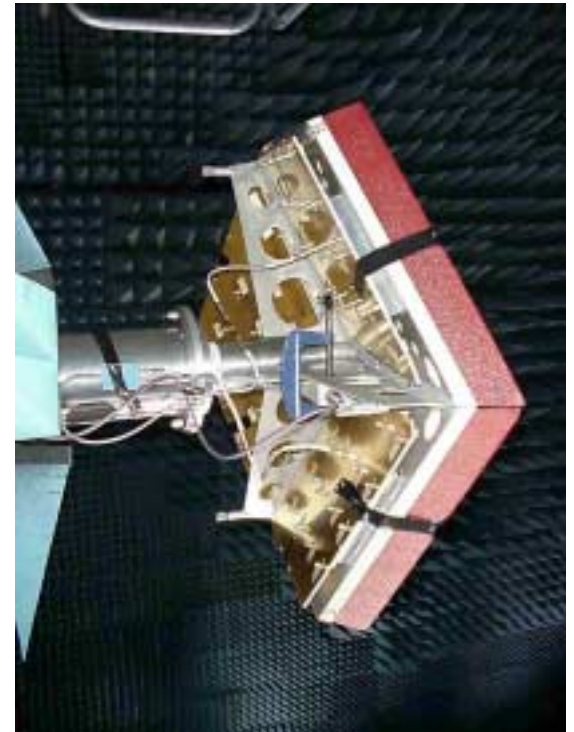
- Data processing will be done by Starlab including analysis of UK-DMC data for altimetric purposes provided by SSTL

PARIS Antenna Breadboard (BNSC NEWTON)

- Antenna equipped with three 16x16 Butler Matrix BFNs. OA diameter ~0.9m
 - 9 COTS helix array elements used per facet
 - 16 beam ports per facet (48 beam total capability)
 - Partial overlap of beam sets from separate array facets allows selective gain enhancement
 - Intended coverage zone is 100° diameter circle
- 



TRIO antenna mounted
on the Portsmouth NFFF
range positioner tower for
radiation pattern
measurements



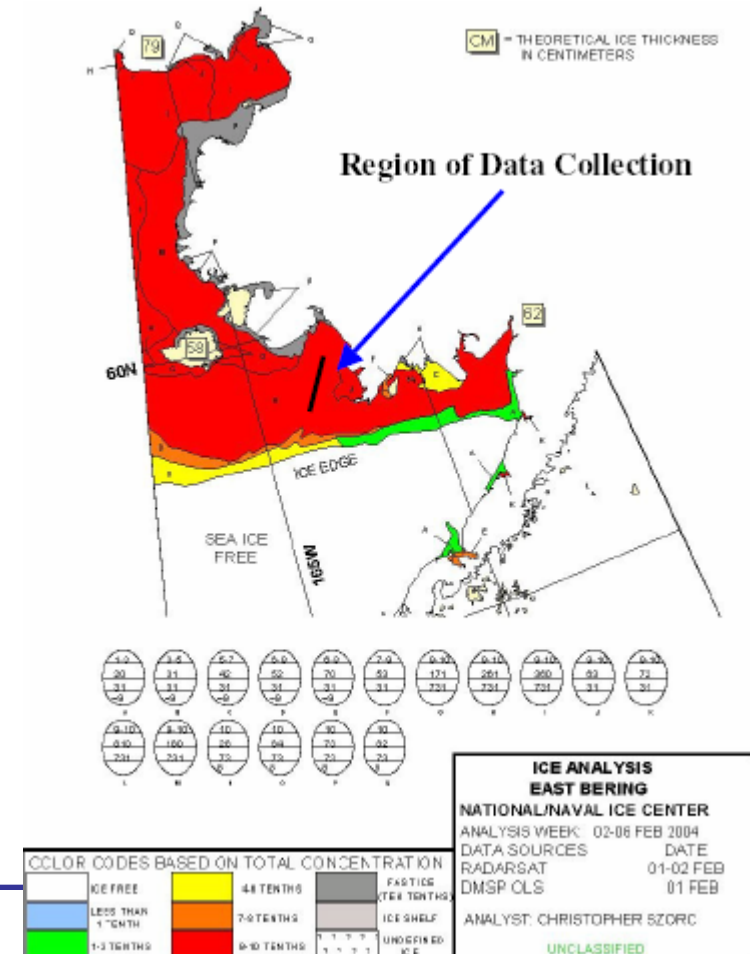


Further PARIS activities:

- Support to SMOS MDPP-3 campaign
 - Provide ocean roughness measurements (MSS)
 - Coast of Norway
 - March/April 2006
- Passenger Instrument on CryoSat ASIRAS Campaign
 - Flights over sea-ice
 - Svalbard
 - Attempt to recover ice reflected GPS signals
 - April/May 2006



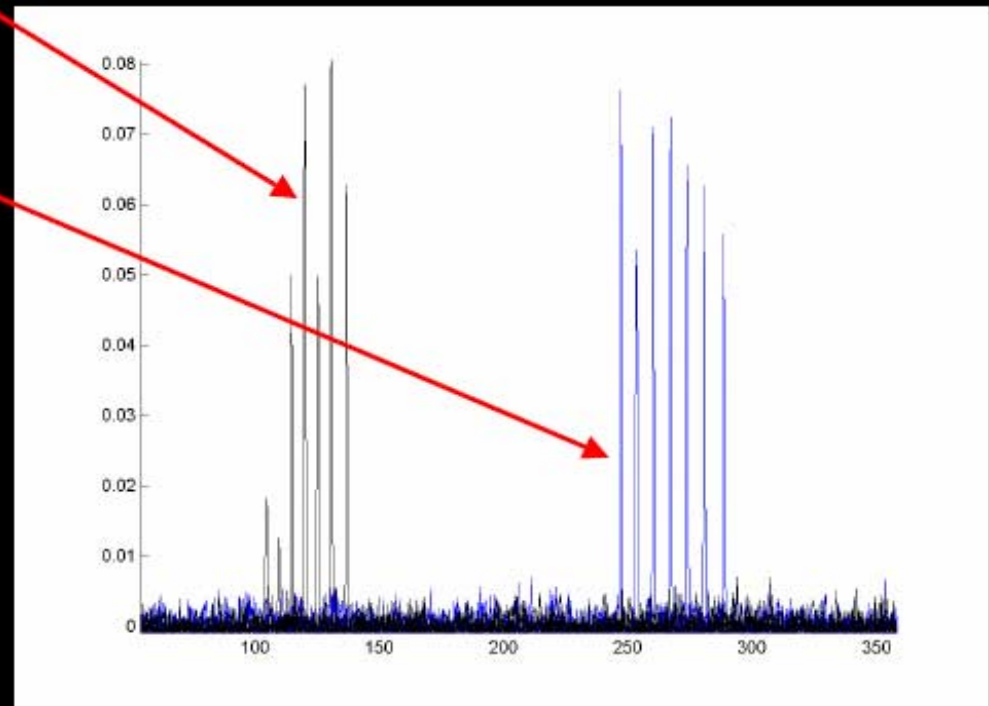
- 4 February 2005
- Data obtained from US National/Naval Ice Centre
 - Point of specular reflection 90-100% covered by ice
 - Total concentration of first year thin ice, 30-70cm thickness
- Unofficially confirmed by phone call to Goodnews Bay





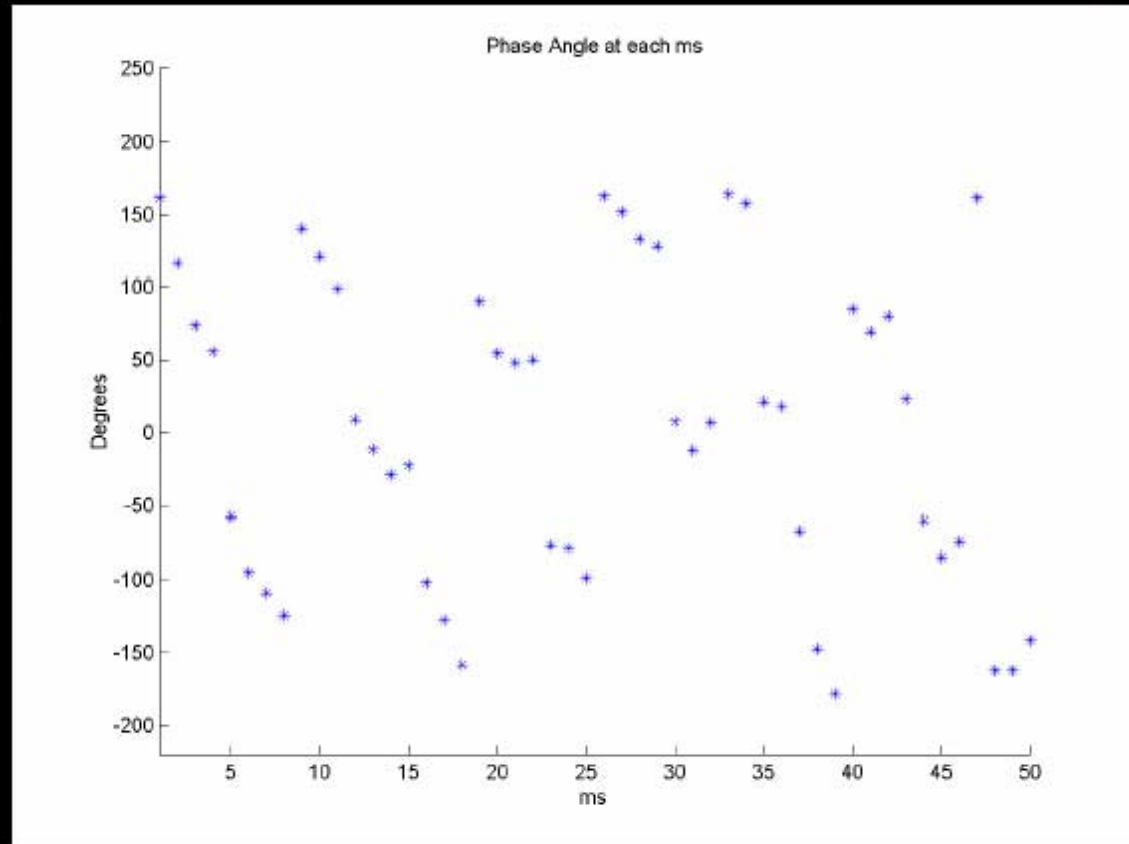
Direct and Reflected Signals C/A Code Phases

- 7 Seconds of data
- Possible Ice Reflections
- Direct Signals at the same time
- Distinctly different
 - ◆ Different Doppler's (>2000 Hz)
 - ◆ Different C/A phase
 - ◆ Different rates of change
- Very Sharp with strong coherent component



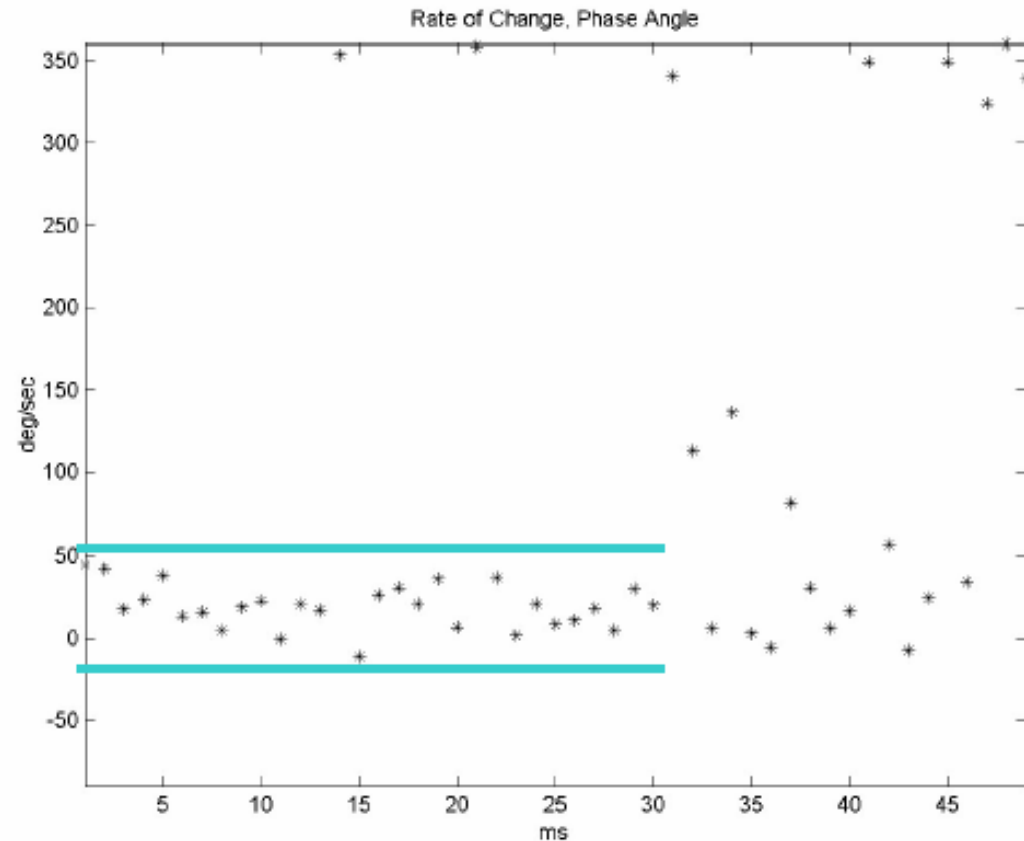


- Raw I and Q Phase Angle Measurements at Consecutive Milliseconds



milliseconds

- Open Loop
- Recoverable Phase
- Observed up to 30th millisecond
- Consistency Deteriorates Near the End of the 50ms

**milliseconds**

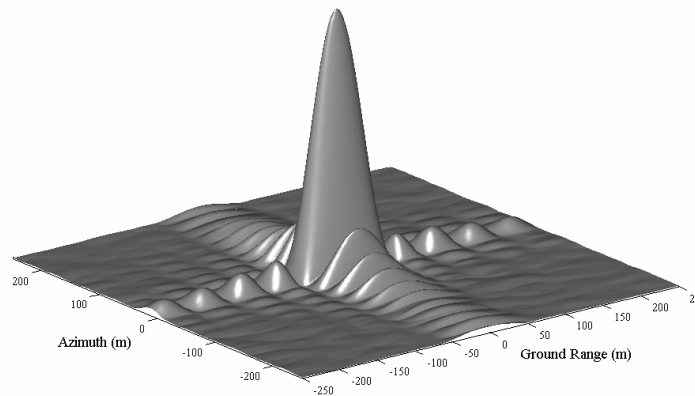


- **7 seconds of UK-DMC data collected over an ice-sheet near Alaska**
- **Both direct and reflected signals detected across entire data set**
- **Magnitude and phase of the reflected signal could be used to sense the ice surface**



ASAR Receiver Gain Droop

- Use digital coded transponder developed by SEA/Qinetiq under ESA contract to repeat ASAR pulses across swath and so measure directly the receiver gain droop of the instrument
- Paper to be presented at EUSAR'06



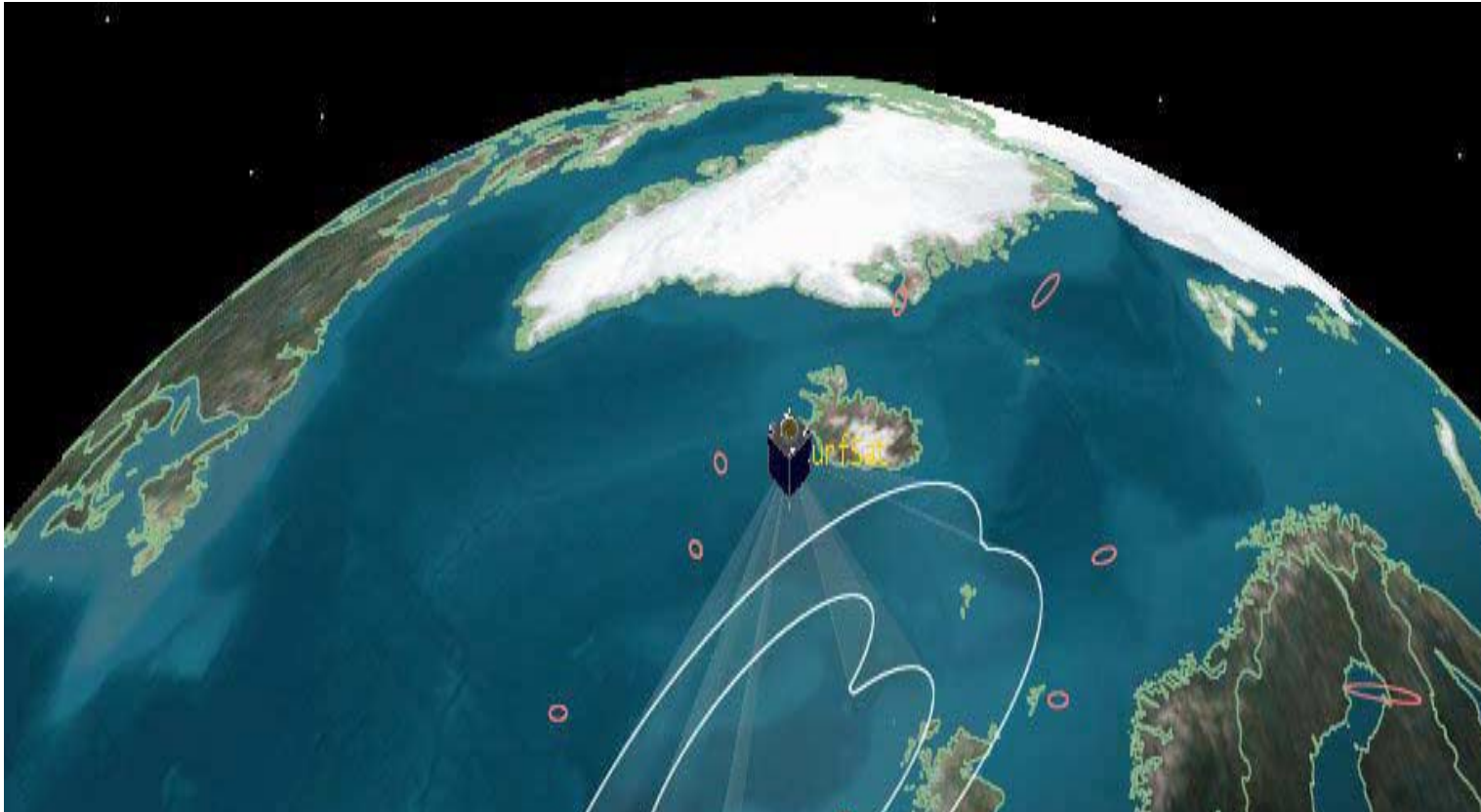


- Based on modified version of SSTL's UK-DMC micro satellite
- Idea to trail or lead SMOS
- Collect GPS ocean reflections from within SMOS swath
- Determine ocean roughness (MSS) for those patches
- Feed results into ocean roughness models
- Improve accuracy of ocean salinity measurements





Indication of coverage





- **Workshop on RF Sensors for Earth Observation**
 - Date: TBD/06
 - Location: ESTEC, The Netherlands
- **Workshop on GNSS Reflections (applications and techniques)**
 - Date: TBD/06, probably June
 - Location: ESTEC, The Netherlands