# Japanese AQ missions and activities

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Dec 2009	GMAP-Asia Passed JAXA mission Definition Review
Jan 2011	APOLLO air pollution mission Selected as top 2 candidate for ISS large class mission by JAXA Earth observation commission
Dec 2013	<b>uvSCOPE</b> Selected as top 2 candidate for ISS middle class mission by JAXA Earth observation commission

#### **Current:**

 Data analysis system for synergy analysis using HIMAARI-8 satellite with several other satellite datasets
 micro-satellite for NO2 hot spot with 1x1 horisontal resolution.

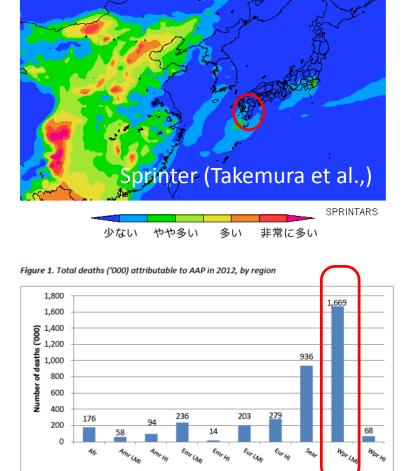


# Motivation

In Kyusu iland, more than 70% of pollution come from outside of Japan.

WHO report (2014): Globally, 3.7 million deaths were attributable to ambient air pollution (AAP) in 2012. The Western Pacific and South East Asian regions bear most of the burden with 1.67 million and 936'000 deaths, respectively. About 236'000 deaths occur in the Eastern Mediterranean region, 200'000 in Europe, 176'000 in Africa, and 58'000 in the Americas. The remaining deaths occur in high-income countries of Europe (280'000), Americas (94'000), Western Pacific (67'000), and Eastern Mediterranean (14'000).

## Serious damage: How about to provide a prediction of detail health index for Fukuoka-city in 1km level?



12:00 JST 10 Oct 2016

PM2.5 prediction

AAP: Ambient air pollution; Amr: America, Afr: Africa; Emr: Eastern Mediterranean, Sear: South-East Asia, Wpr: Western Pacific; LMI: Low- and middle-income; HI: High-income.

## Project

A Pollution prediction system in Fukuoka, Kyushu, Japan

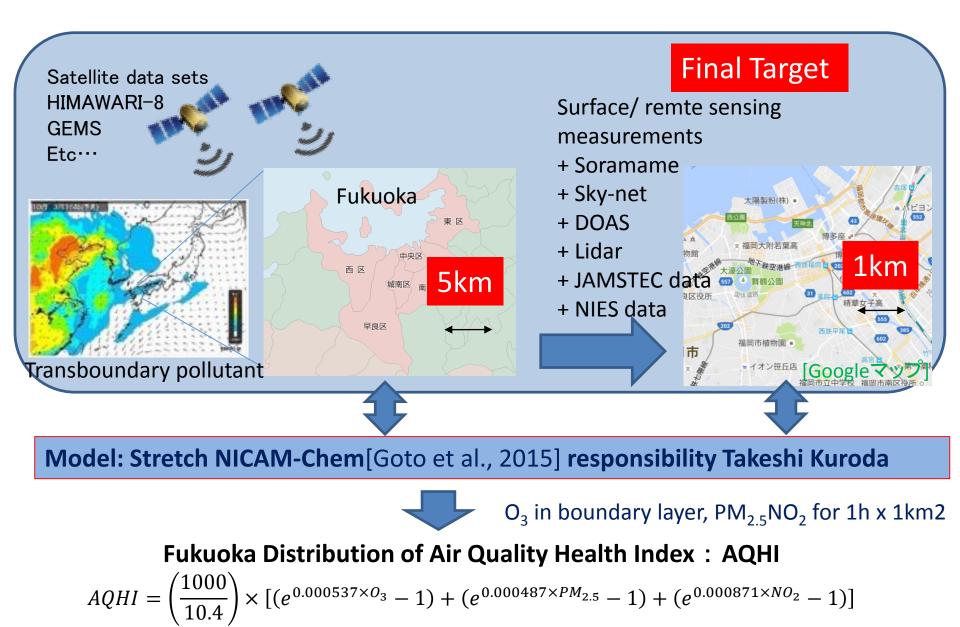


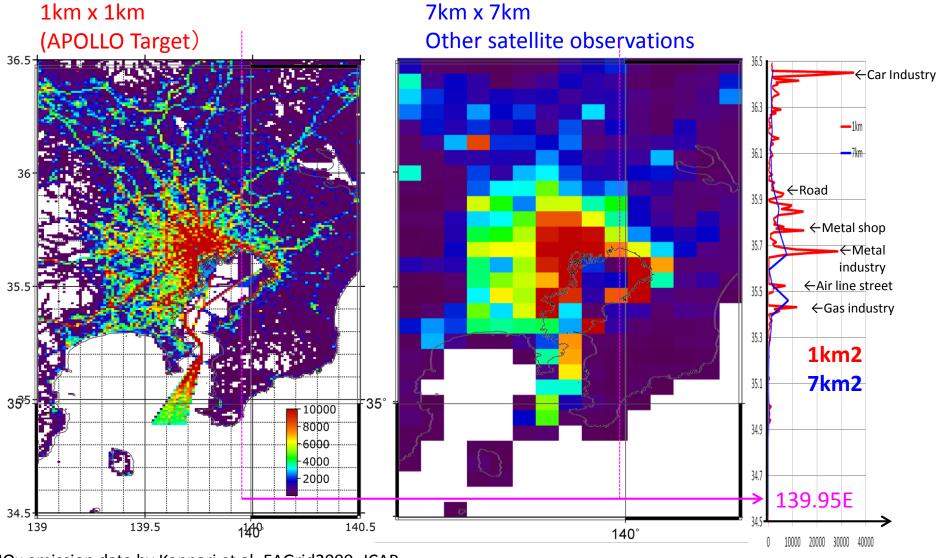
In Kyusu iland, more than 70% of pollution come from outside of Japan.

There are many ground-based observation base for pollution species.

## Project

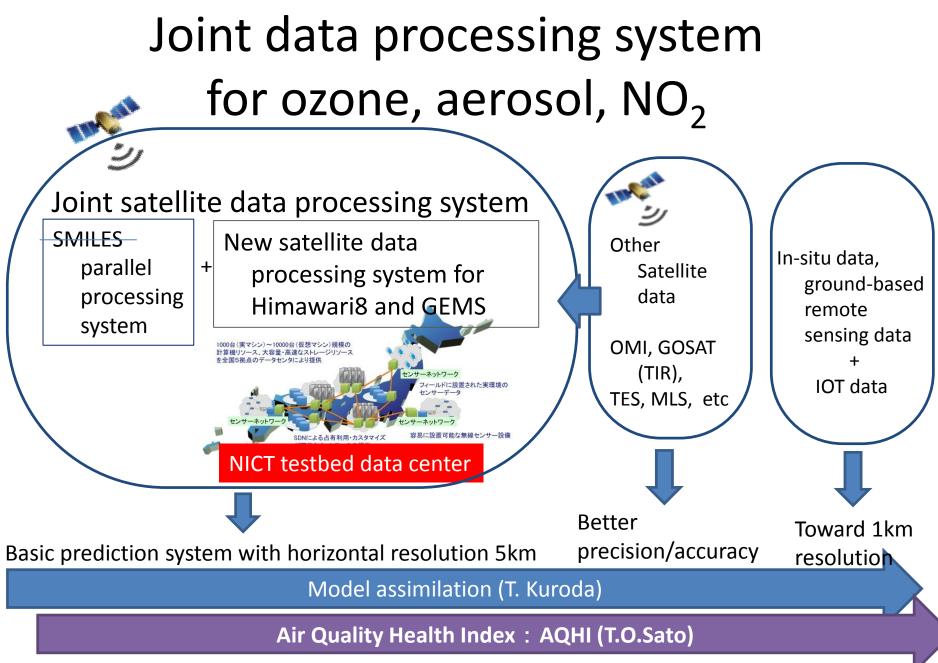
# A Pollution prediction system in Fukuoka, Kyushu, Japan



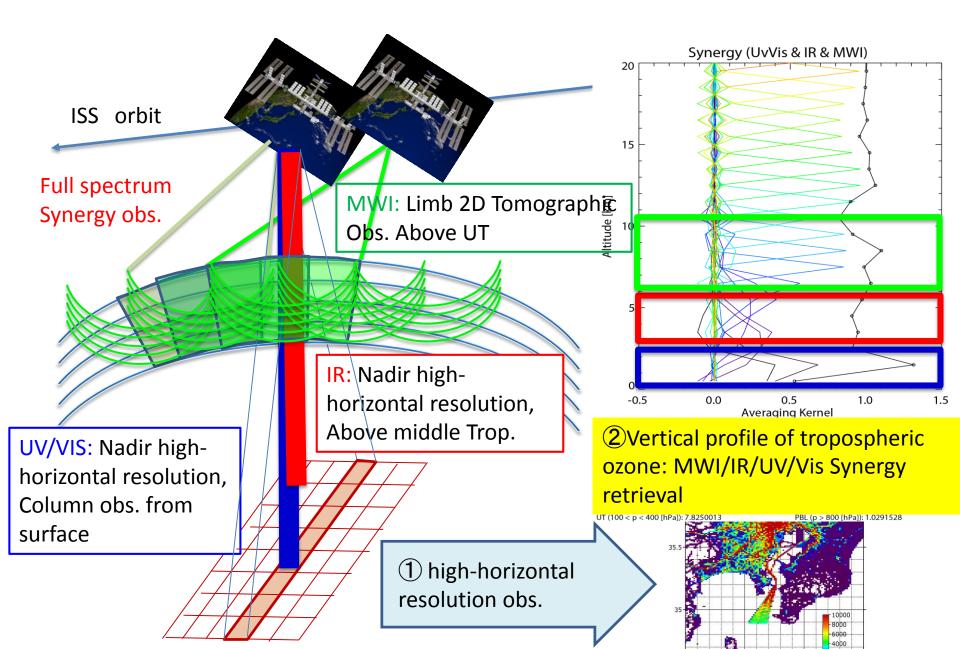


NOx emission data by Kannari et al., EAGrid2000, JCAP

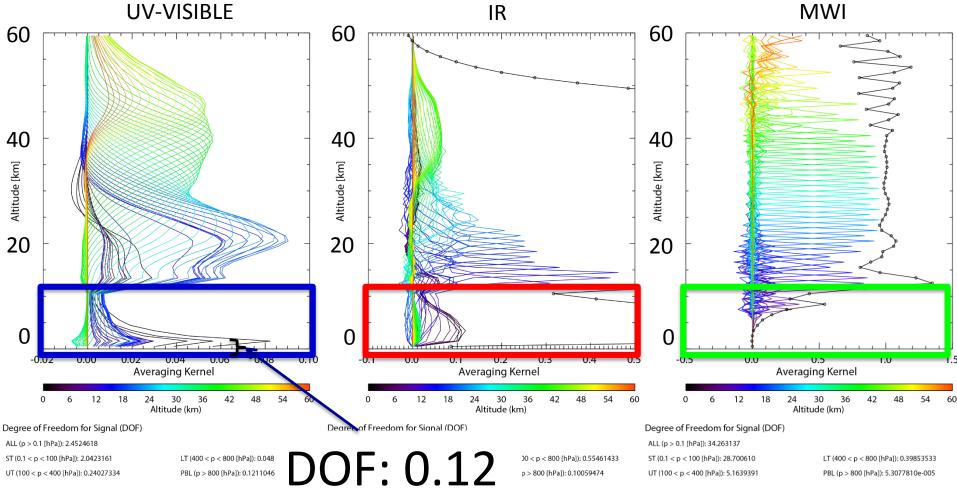
#### Responsibility T. O. Sato



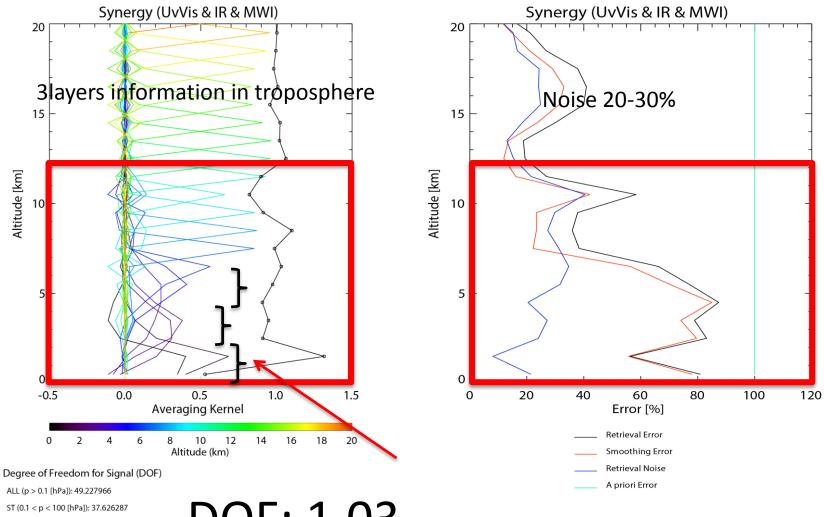
## synergy observation



## **O3** Averaging Kernel



# **O3** Averaging Kernel



UT (100 < p < 400 [hPa]): 7.8250013

DOF: 1.03

#### **Current:**

**1.** Data analysis system for synergy analysis using HIMAARI-8 satellite with several other satellite datasets

2. micro-satellite for NO2 hot spot with 1x1 horizontal resolution.



Mission concept: A microsatellite with high horizontal resolution

### In this study, we present a feasibility study for observation precisions of NO<sub>2</sub>

- $\rightarrow$ Synthetic spectra are calculated using SCIATRAN (RTM) with two geometries.
- →Slant column densities (SCDs) are derived by DOAS method and converted to vertical column densities (VCDs).

### Scientific requirement

Products	Standard: NO <sub>2</sub> ,O <sub>4</sub>		
Detection limit ( <u>NO<sub>2</sub> column density</u> )[molecules/cm <sup>2</sup> ]	<pre><u>3.0 x 10<sup>15</sup> (</u>5%) (about 0.6ppb in boundary layer)</pre>		
(ex) Tropospheric NO <sub>2</sub> column amount [molecules/cm <sup>2</sup> ]	6.0 x 10 <sup>16</sup> (Boundary layer 4.7 x 10 <sup>16</sup> )		
IFOV	1 km x 1km (2 km x 2 km)		
Vertical resolution	Tropospheric column		
Swath width	approx. 200 km		

## 今後のきぼう曝露部ミッションの募集区分と募集分野

募集区分は大きく3つ。それぞれ以下の分野のミッションを募集。

①大型ミッション: 船外ポート利用/船外打上

「こうのとり(HTV)」などの船外貨物として打ち上げる必要のある、 概ね1m<sup>3</sup>以上の大型の船外利用ミッション

#### く募集分野>

▶ 地球観測(例:地球科学、観測センサ、データ処理技術)

- ▶ 宇宙科学(例:科学観測、惑星科学、工学・技術実証)
- ▶ 技術開発(例:ロボティクス、エネルギー伝送)

②中型ミッション: 船外ポート利用/船内打上

船内貨物として打ち上げ、きぼう船内からエアロックを使って船外 に出し、ロボットアームで船外ポートに取り付ける利用ミッション

<募集分野> 同上(地球観測、宇宙科学、技術開発)

③小型ミッション: 簡易船外利用/船内打上

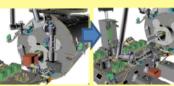
船内貨物で打ち上げ、きぼう船内からエアロックで船外に出し、 エアロックに取り付けたまま、またはロボットアームで把持したまま、 もしくは船外のハンドレール等に取り付けて利用するミッション

#### <募集分野>

地球観測、宇宙科学、技術開発、物質·物理科学、 生命科学、宇宙医学



船外貨物として打上げ、そのまま船外ポートへ取り付けて利用 例:MAXI(X線観測): 寸法0.8m×1.0m×1.85m、重量約500kg





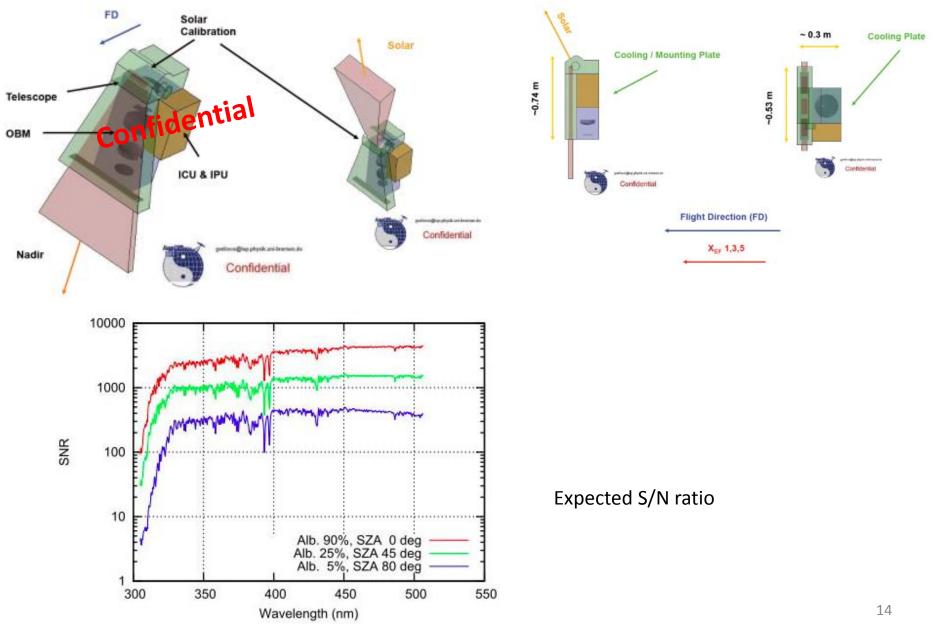
船外へ移動

の子アームで取付 て運用

中型ミッション: 寸法37×80×48[cm]程度、重量約80kg以下



# **UV/VIS Sensor**

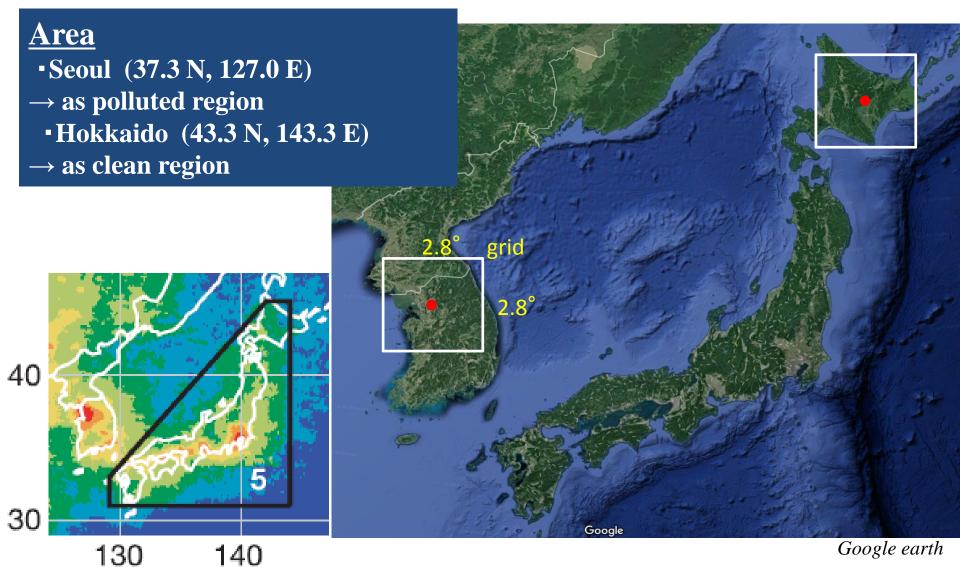


## Parameter setup in this study

Parameters	The number of	Elements of parameters		
	parameters			
Wavelength range	2	Reference : 425 - 450 (conventional region) 460-490		
Area	2	Seoul (as polluted area) Hokkaido (as clean area)		
Season	1	Winter(Jan.)		
Altitude of satellite	1	300 km		
Spectral resolution (FWHM)	1	0.4 nm		
Sampling step	1	0.1 nm		
Detector size	1	0.064 mm		
Aperture size	1	0.74 cm		
Read noise	1	10		
IFOV	2	1 x 1 km <sup>2</sup> 2 x 2 km <sup>2</sup>		

### Setup of each parameter

# Input profile data - Area selection



Andreas et al., (2005) nature

# **Results – Expected SNRs for each IFOV case**

#### SNRs expected in each wavelength and spatial resolution

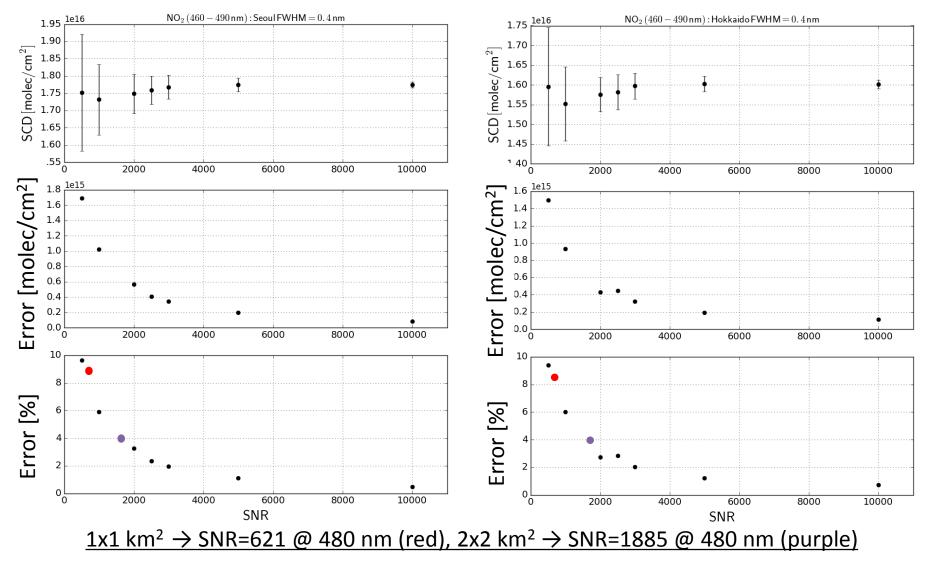
Wavelength [nm]	Spatial resolution [km <sup>2</sup> ]	Expected SNR (Electronics)	Expected SNR (Shot)	Dark current SNR	Expected SNR (all)
480	1x1	1642	691	2806	621
(460 - 490)	2x2	9287	1954	11222	1885
450	1x1	1411	620	2518	554
(425 - 450)	2x2	7983	1753	10073	1688

•We assumed that the detector was Si-CMOS 2D array sensor and the optical efficiency and quantum efficiency was 0.43.

• In both cases of wavelength, expected SNRs of  $2 \times 2 \text{ km}^2$  were about three times larger than them of  $1 \times 1 \text{ km}^2$ .

# **Results and discussion**

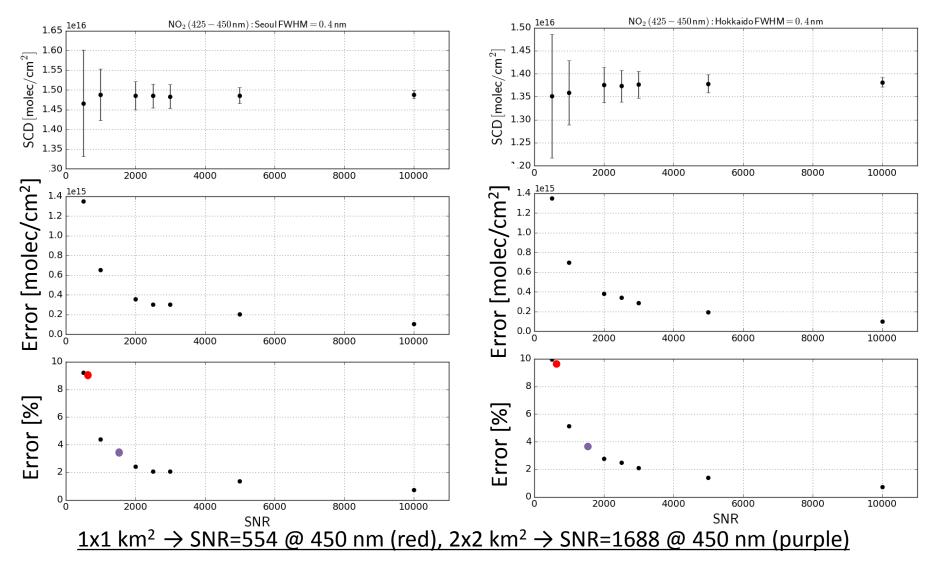
Winter (Jan.), Alt. of satellite = 300 km, FWHM = 0.4, 460-490 nm,  $\Delta\lambda$  =0.1 nm Left : Seoul Right : Hokkaido



Upper figure : SCD (total = tropo. + strato.), Middle figure : Error [molec/cm<sup>2</sup>], Lower figure : Error [%]

# **Results and discussion**

Winter (Jan.), Alt. of satellite = 300 km, FWHM = 0.4, 425-450 nm,  $\Delta\lambda$  =0.1 nmLeft : SeoulRight : Hokkaido



Upper figure : SCD (total = tropo. + strato.), Middle figure : Error [molec/cm<sup>2</sup>], Lower figure : Error [%]

#### **Current:**

**1.** Data analysis system for synergy analysis using HIMAARI-8 satellite with several other satellite datasets

- 2. micro-satellite for NO2 hot spot with 1x1 horizontal resolution.
- Scientific requirement of NO2 observation is about 5%
- Spatial resolution of 1x1 km<sup>2</sup> provided SNR about 500 and the measurement precision to be 9% in this simulation.
- Spatial resolution of 2x2 km<sup>2</sup> provided SNR about 1700, and a measurement precision was about 4%.
  - further lower orbit
  - change the detectors
  - more broader frequency resolution



Current satellite horizontal resolution is about 7-25km, which is not enough to estimate the inventory from the "large stationary emission source", such as transport sector (car, ship), agriculture region, industrial region, and city.



ISS altitude will give higher horizontal resolution than any of past/future missions  $Current 7-25 \text{ km} \rightarrow APOLLO 1-2 \text{ km}$ 

