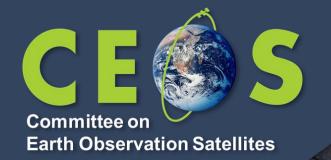
# LSI-VC-18

JAXA Mission Updates



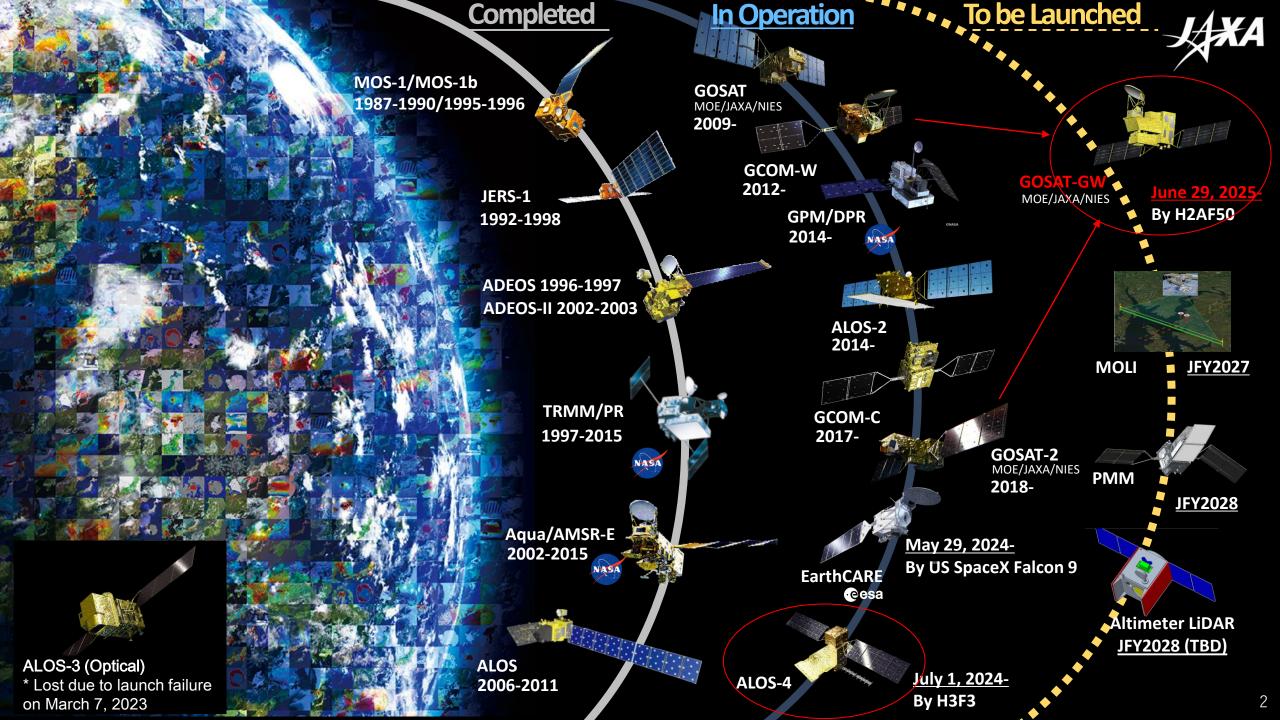
Takeo Tadono, JAXA

Agenda Item 3.7

LSI-VC-18

Ispra, Italy

**2-5 September, 2025** 



# GOSAT-GW: Global Observation SATellite for Greenhouse gases and Water cycle



# AMSR3 (Advanced Microwave Scanning Radiometer 3)



AMSR3, led by JAXA, will succeed AMSR series observations adding new high frequency channels for solid precipitation retrievals and water vapor analysis in NWP



GOSAT-GW at Tanegashima Space Center on 20 May 2025

# TANSO-3 (Total Anthropogenic and Natural emissions mapping SpectrOmeter-3)

TANSO-3, led by Japanese Ministry of the Environment (MOE) and National Institute of Environment Studie (NIES), will improve observation capability of greenhouse gases from GOSAT-2/TANSO-2

The GOSAT-GW was successfully launched on 29<sup>th</sup> June 2025 (JST) by H-IIA F50 rocket (final flight)



#### **GOSAT-GW Satellite Specifications**

Mission Instruments		AMSR3 (JAXA) TANSO-3 (MOE/NIES)	
	Туре	Sun-synchronous, Sub-recurrent orbit	
Orbit	Altitude	666km, recurrent cycle 3days (same as GOSAT)	
	Local sun time at ascending	13:30±15min (same as GCOM-W)	
	Revisit time	3 days	
Satellite Mass		2.6 tons (including propellant)	
Designed lifetime		> 7 years	
Launch		<b>29</b> June <b>2025</b> (JST) by H-IIA F50	



# Mission Targets and AMSR3 Products



#### Aims of AMSR3

- Succeed AMSR series' observation of water-related parameters
- Respond to new user needs by hardware/software improvements



**AMSR** 

Aqua

2002.5-2011.10

**AMSR-E** 



GCOM-W

2012.5-present

AMSR2



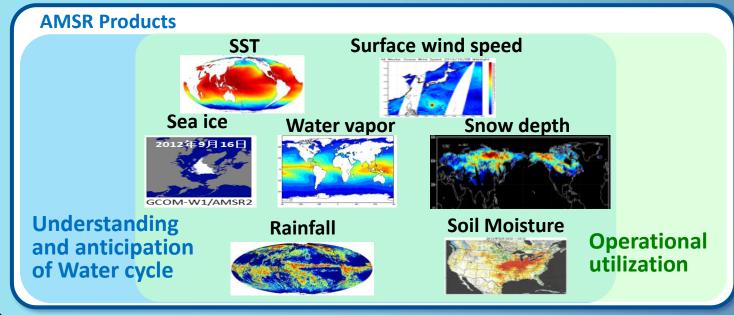
GOSAT-GW

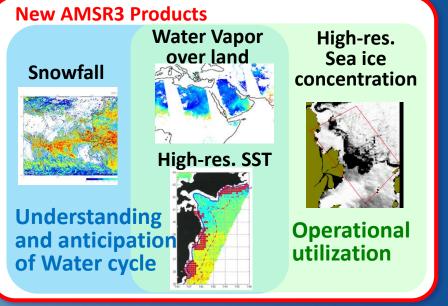
**2025.6-present** 

AMSR3

New stakeholder's needs

#### **AMSR3 products**



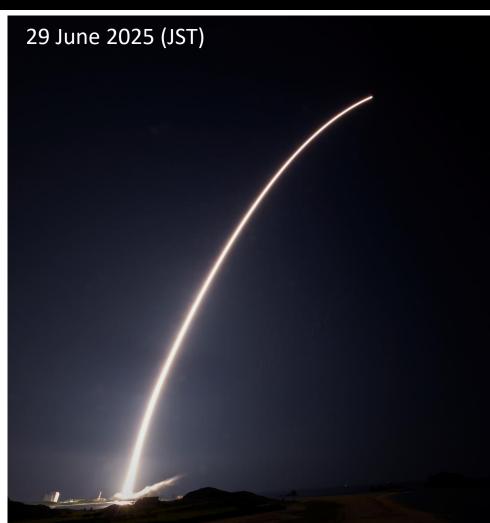


# Launch of the GOSAT-GW by H-IIA F50







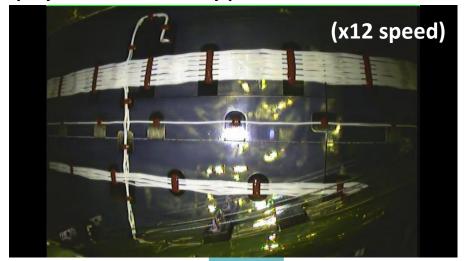




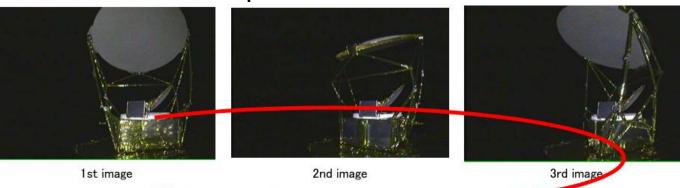
# Results of Critical Operations Phase

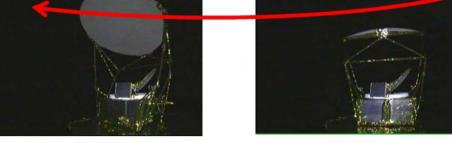


#### Deployment of solar array paddle #1 at 2:25JST on 29 June



#### Initial run-up of AMSR3 at 19:45JST on 30 June





th image 4th image

Antenna drive mechanism of AMSR3 is slowly rotated (4rpm) to confirm that the angular momentum reaches the set target value.

#### Deployment of AMSR3 antenna at 14:57JST on 30 June





#### **Completed Critical Operations Phase on 1 July**

The satellite is in a stable condition to remain in orbit, and moved to Initial Functional Verification Operations Phase.



# Mission Schedule after the Launch



29 June 2025 Launch

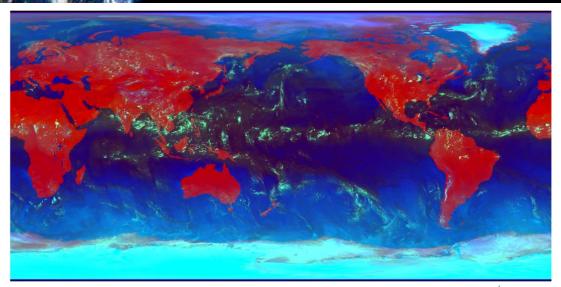
After the satellite separation, Deployment of the solar array paddles, antenna deployment & initial run-up of AMSR3, release of the pointing mirror lock for TANSO-3  Completed on 1 July  Completed on 1 July  Completed on 1 July  Calibration and validation of AMSR3 and TANSO-3  Early data release to partner agencies and selected principal investigators is planned during this period.  Tender of AMSR3 standard product to public *TANSO-3 products will be released by MOE/NIES.  Tansonth after the launch  12-month after the launch	Critical Operations Phase	Initial Functional Verification Operations Phase	Initial Calibration and Validation Operations Phase	Nominal Mission Operations Phase
i i	Deployment of the solar array paddles, antenna deployment & initial run-up of AMSR3, release of the pointing mirror lock for TANSO-3	satellite and onboard instruments	Early data release to partner agencies and selected principal investigators is planned during this period.	product to public *TANSO-3 products will be

7-year after the launch

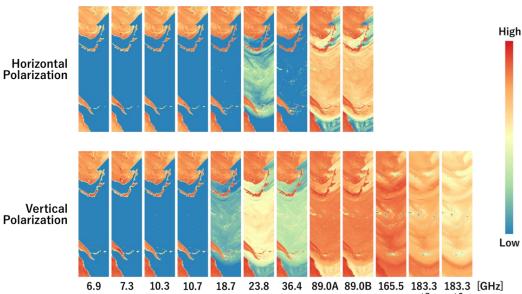


# Early Observation Results of AMSR3 released on 5<sup>th</sup> Sep 2025





\* https://global.jaxa.jp/press/2025/09/20250905-1\_e.html

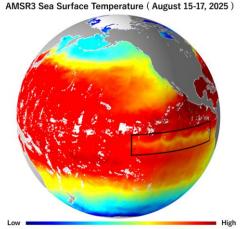


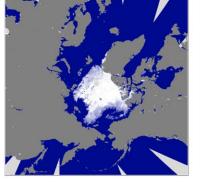
Brightness temperature of 11 ch measured by AMSR3. (Aug. 16, 2025)

JAKA AMER Brightness temperature of 18.7, 89.0, 165.5 GHz V pol. by AMSR3

(Aug. 15-17, 2025)

Sea Surface temperature (SST) derived by AMSR3. (Aug. 15-17, 2025)





AMSR3 Sea Ice Concentration August 15, 2025 Ascending



Sea Ice distribution in north pole (left) and south pole (right) derived from AMSR3. (Aug. 15, 2025)

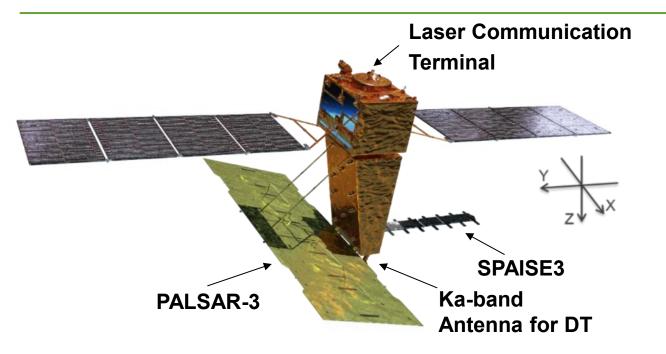
AXA AMER



JAKA AMER

### **ALOS-4 Overview**





Same orbit as ALOS-2

Lack properties of the second properties of the s

SPAISE3: SPace based Automatic Identification System Experiment 3 PALSAR-3: Phased Array type L-band Synthetic Aperture Radar-3

Designed Life Time		7 years	
Mission Instruments		PALSAR-3, SPAISE3	
Size (X, Y, Z)		10.0m×20.0m×6.4m	
Satellite Mass		Approx. 3,000 kg	
Electri Solar Array		Approx. 7,200 W	
city	Battery	380 Ah	
Data Recorder		Approx. 1 Tbyte	
		Direct Trans. by Ka-band 3.6 Gbps	
Data Tra	nsmission	Via. Laser Utilizing Communication	
		System (LUCAS) by 1.8 Gbps	
	Туре	Sun-synchronous	
Orbit	Altitude	628 km	
	LSDN	12:00	
	Revisit Cycle	14 days	
	Inclination	97.9 deg.	

# **ALOS-4 Operation Status**



- Launched by H3 Launch Vehicle flight No. 3 on <u>July 1, 2024</u>.
- The Initial Functional Verification operation was completed on October 18, 2024, and the satellite moved to the Initial Calibration and Validation (ICV) operation.
- The ICV operation was scheduled to three months, but it was decided to continue until the end of March 2025 because it took more time than expected to adjust the image quality for the world's first on-orbit demonstration of digital beam forming (DBF).
- The ICV was completed on <u>March 31, 2025</u>, and the Nominal Observation operation started from <u>April 1, 2025</u>. The data distribution was started from July 2025.





**ALOS-4** 

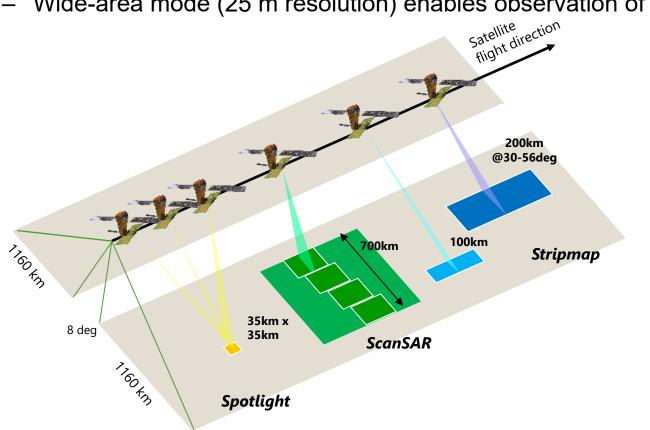
H3F3 launch

July-Sep. 2024		-Dec. 2024	JanMarch 2025	April 2025~
Initial Functional Verificatio		Initial Cal/Val (ICV) operation		Nominal Observation operation
▲Launch On July 1 <sup>st</sup> 2024				▲Start of provision of PALSAR-3 standard product
				▲Start of the Basic Observation Scenario (BOS) and emergency observation

### **ALOS-4 PALSAR-3 Overview**



- Expansion of Observation Swath and Improvement of Observation Frequency
  - New technologies such as Digital Beamforming (DBF) enable observation of a 200 km swath in high resolution 3 m mode (50 km swath in ALOS-2) and observation of the entire Japan about 20 times a year (approx. 4 times in ALOS-2).
  - Wide-area mode (25 m resolution) enables observation of 700 km width (standard 350 km for ALOS-2)



70 deg

#### Observation swath width

Modes	ALOS-4	ALOS-2	
Stripmap (res. 3/6/10 m)	100-200 km	30-70 km	
ScanSAR (res. 25 m*)	<u>700 km</u>	350-490 km	
Spotlight (res. 1 x 3 m)	<u>35km</u> <u>×35km</u>	25km ×25km	

<sup>\*</sup> Single look

### **PALSAR-3 Calibration Status**



### ■ Progress of calibration and validation of each observation mode and beam as of July 2025

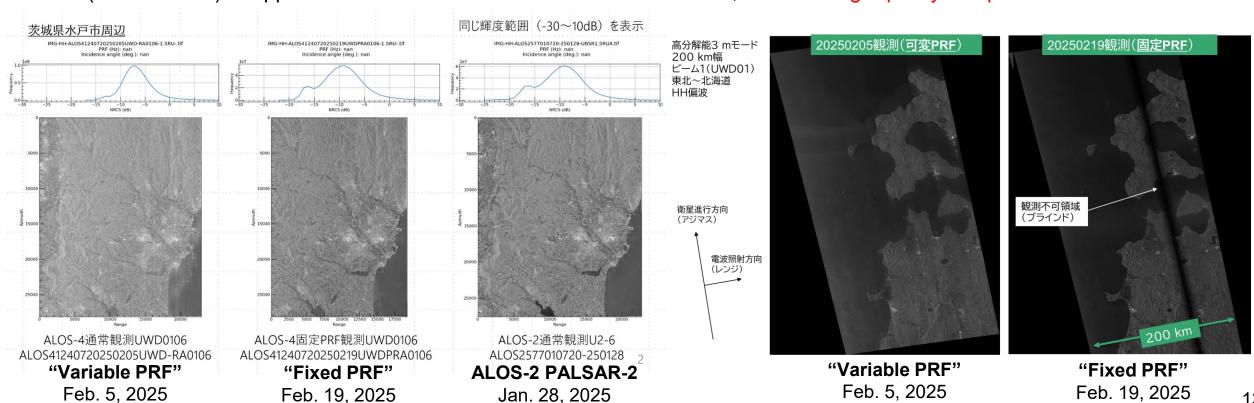
Calibration plan	Beam code	Observation mode	Purpose	Status
Initinal cal./val. phase	SBD	Spotlight	Emergency/individual observation	Completed
	UWD01~03	Stripmap 3 m/200 km swath		Completed
	HBQ06, 08	Stripmap 6 m/100 km swath (full-polarimetry)	Basic observation,	
	FWD01	Stripmap 10 m/200 km swath	emergency/individual observation	·
	XBD02	ScanSAR		
	UBD01~05, UBD14~23	Stripmap 3 m/100 km swath	Emergency/individual observation	On-going
	UBQ05~09	Stripmap 3 m/100 km swath (full-polarimetry)	Same as above	On-going
Regular observation phase	HBQ05, 07, 09	Stripmap 6 m/100 km swath (full-polarimetry)	Same as above	On-going
	FWD02~03	Stripmap 10 m/200 km swath	Same as above	Completed
	XBD01, 03	ScanSAR	Same as above	On-going
	FWD01	Stripmap 10 m/200 km swath (ionospheric correction on-board)	Basic observation, emergency/individual observation	Pending (operational constraints)
No calibration plan	Others	Others	No plan for use	N/A

<sup>\*</sup> https://www.eorc.jaxa.jp/ALOS/en/alos-4/a4\_calval\_e.htm

# PALSAR-3 Initial Cal/Val Operation Results



- Azimuth ambiguity (false images) in some areas
- In the DBF-SAR technology, which will be newly developed in ALOS-4, it has been confirmed that many false images occur in the satellite flight direction (i.e., azimuth direction) in some areas when operating with the "Variable PRF (pulse repetition frequency)" setting, a function to create seamless images. Image quality adjustment is necessary continuously and is scheduled to be taken care of during FY2025.
- To avoid the influence of this phenomenon, the "Fixed PRF" setting may be used in some cases. In this case, an unobservable area (called "blind") of approx. 10-20 % will occur in the observation width, but the image quality is equivalent to ALOS-2.



## PALSAR-3 Basic Observation Operation Plan



■ Basic Observation Scenario (BOS) in the World

