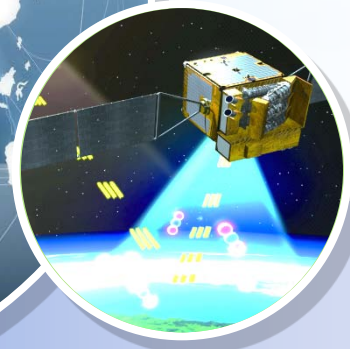


TANSAT MISSION

**GLOBAL CARBON DIOXIDE
OBSERVATION AND MONITORING**



Shanghai Engineering Center
for microsatellites

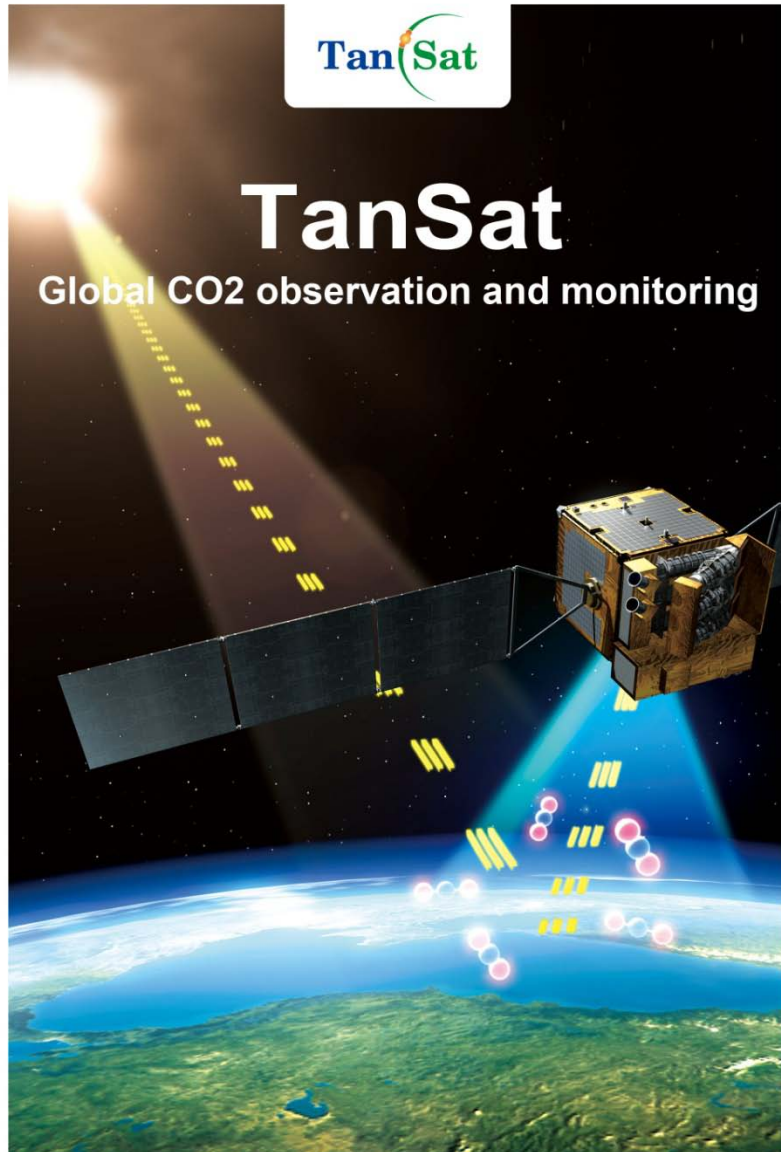
2013.5



Outline

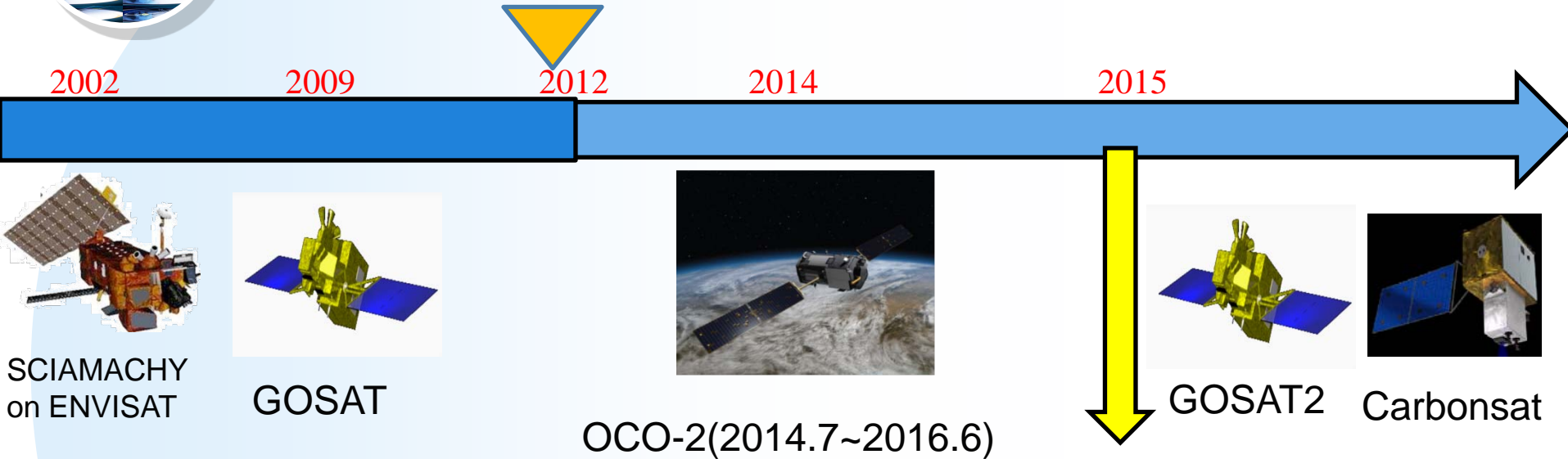
- ❖ **Mission Background**
- ❖ **Scientific Instruments**
- ❖ **Observation Modes**
- ❖ **Cal/Val**
- ❖ **Current Status**
- ❖ **Development Schedule**

Mission Objective



To retrieve the atmosphere column-averaged CO₂ dry air mole fraction (XCO₂) with precisions of 1% (4ppm) on national and global scales.

Greenhouse Observation Missions



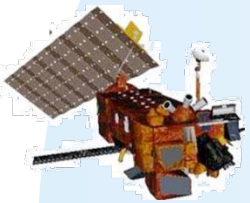
2002

2009

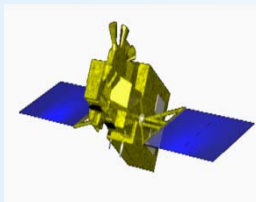
2012

2014

2015



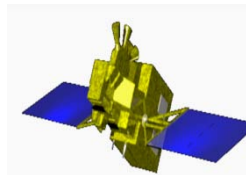
SCIAMACHY
on ENVISAT



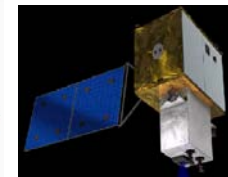
GOSAT



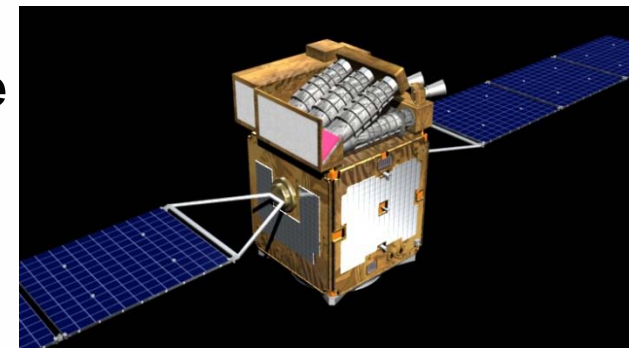
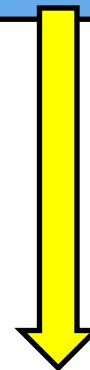
OCO-2(2014.7~2016.6)



GOSAT2

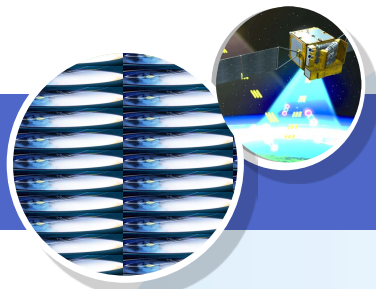


Carbonsat



TanSat(2015.7~2018.6)

- **Founded by NRSCC, The project started at the beginning of 2011**
- **Tansat is planed to be launched in 2015**

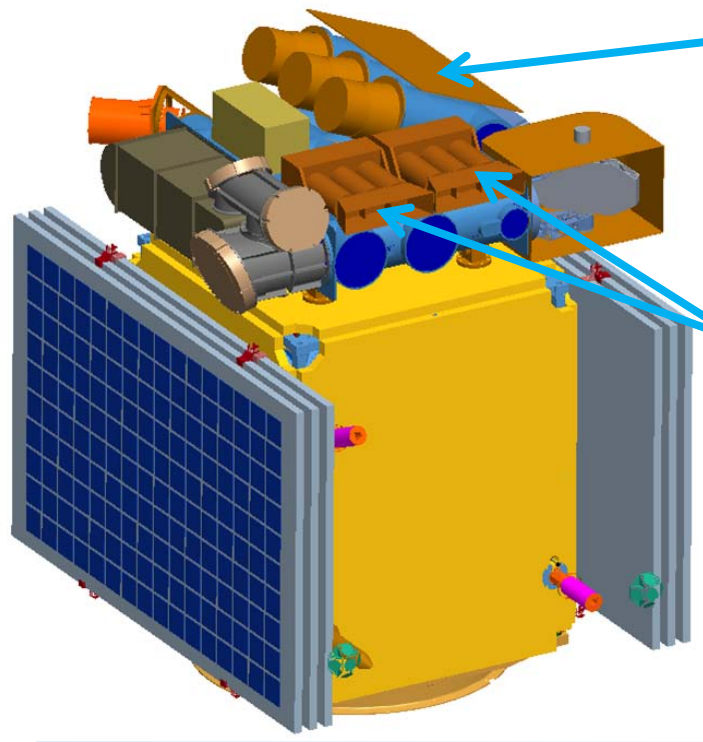


Payload of TanSat

CO2 Spectrometer

Cloud and Aerosol Polarimetry Imager (CAPI)

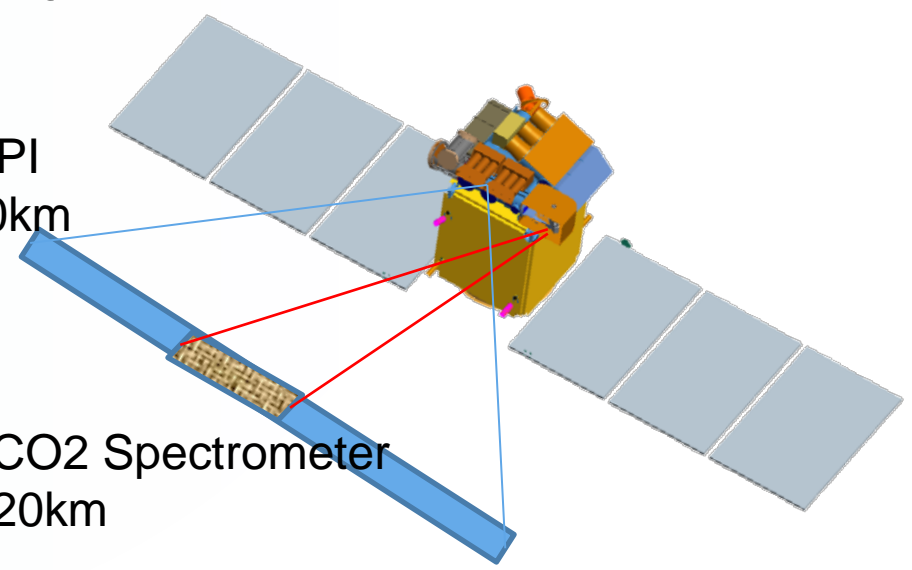
A wide field of view moderate resolution imaging spectrometer with polarization channel

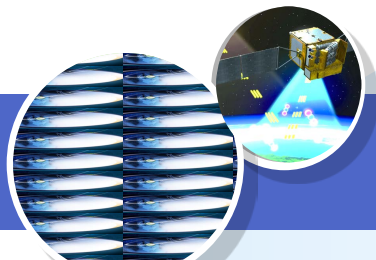


Platform

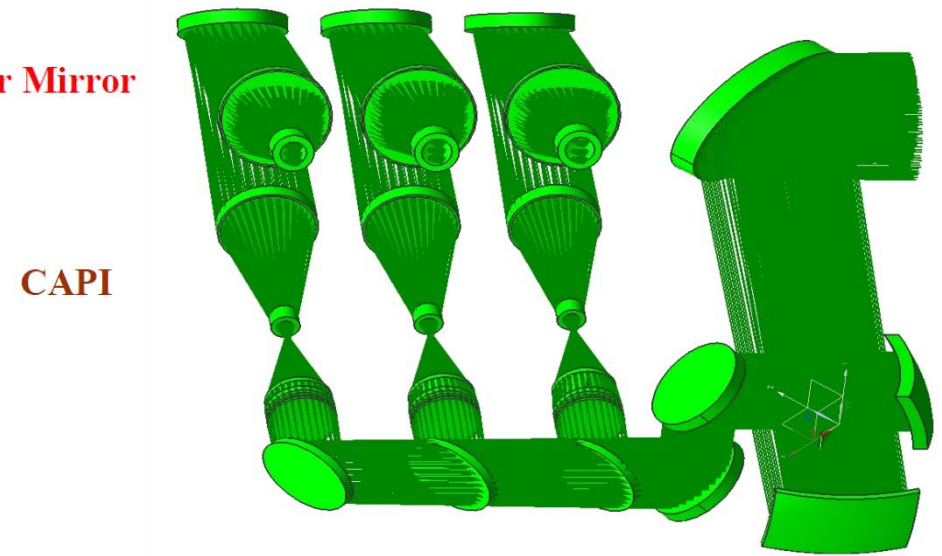
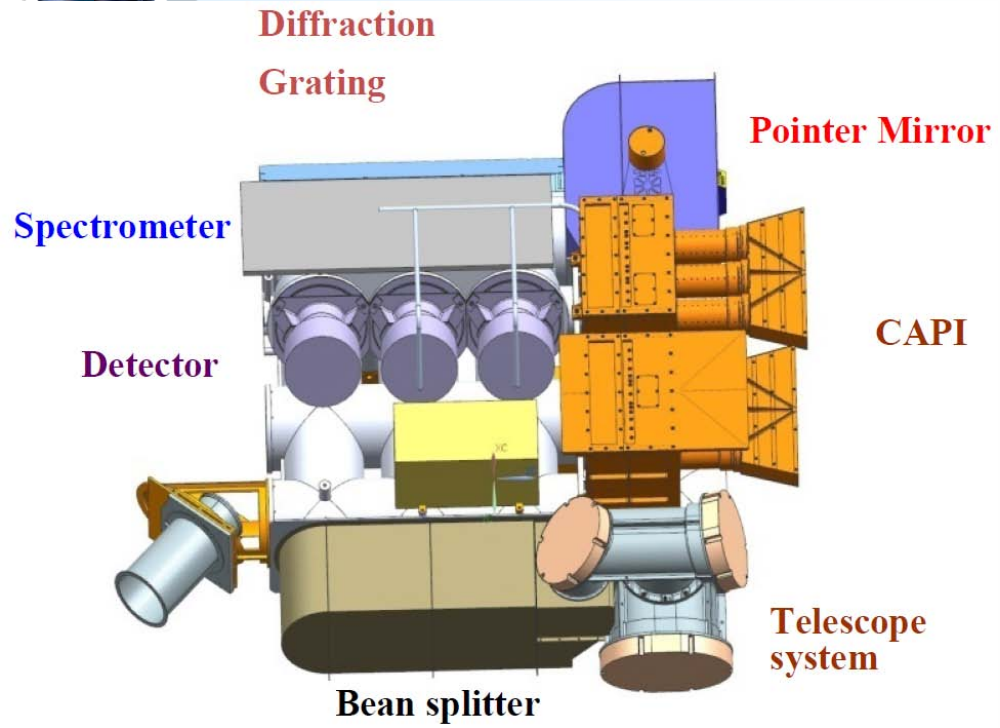
CAPI
400km

CO2 Spectrometer
20km

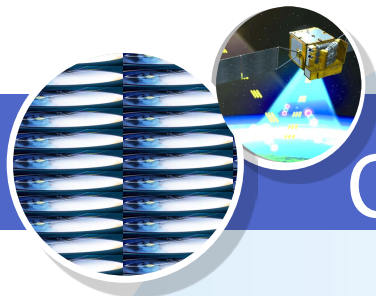




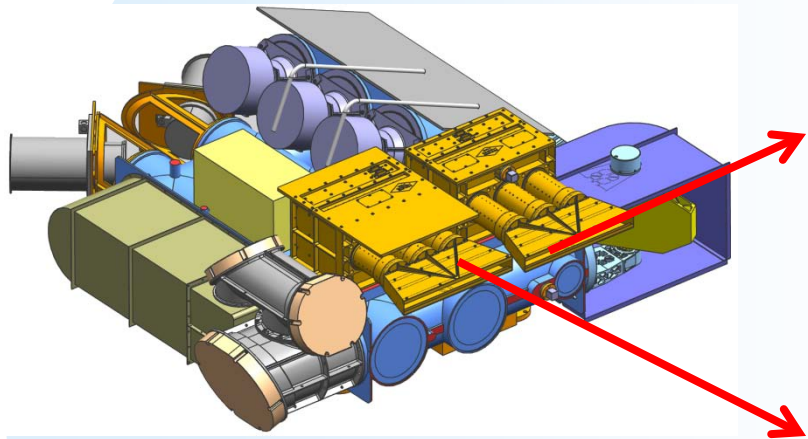
CO2 Spectrometer



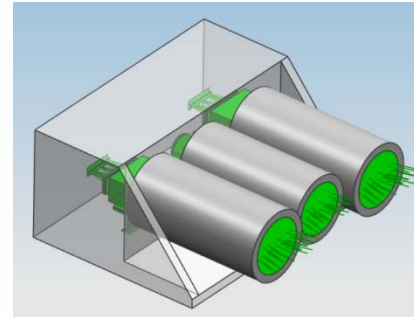
	O2-A	CO ₂ weak	CO ₂ Strong
Spectral Range(nm)	758-778	1594-1624	2042-2082
Spectral Resolution	0.044	0.081	0.103
SNR	360	250	180
Spatial Resolution	1km×2km, 2km×2km		
Swath	20km		



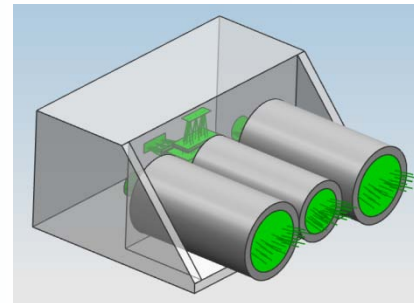
Cloud and Aerosol Polarimetry Imager



VNIR



SWIR



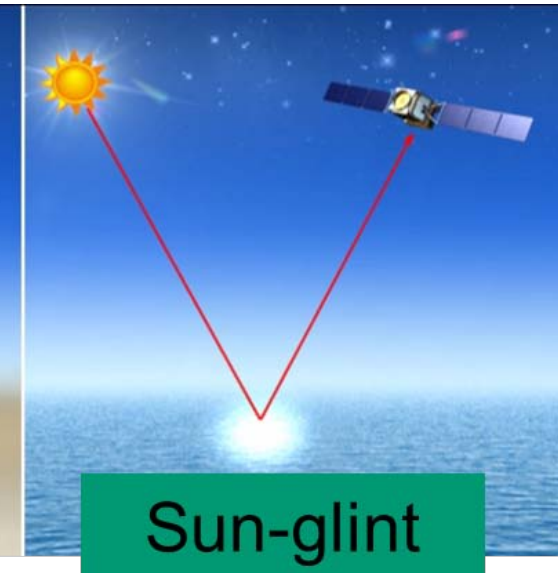
1	0.38 μ m
2	0.67 μ m _{0°} 、 0.87 μ m
3	0.67 μ m _{60°} 、 0.67 μ m _{120°}
4	1.375 μ m、 1.64 μ m _{0°}
5	1.64 μ m _{60°}
6	1.64 μ m _{120°}

	Band (nm)	SNR	Polarization (deg)	FOV	Pixel Num.
1	365-408	260		400× 1km ²	1600
2	660-685	160	0/60/120		1600
3	862-877	400			1600
4	1360-1390	180			800
5	1628-1654	110	0/60/120		800

Observation Modes

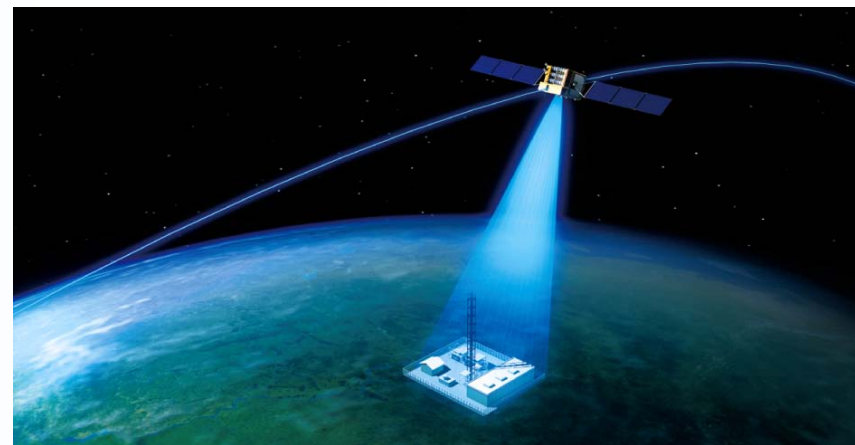
❖ Nadir mode

- land observation
- solar zenith angle < 80 deg
- Principle plane track



❖ Sun-glint mode

- Ocean observation
- solar zenith angle < 70 deg
- Principle plane track



Cal/Val

❖ Specification

- 5%(absolute), 3%(relatively)

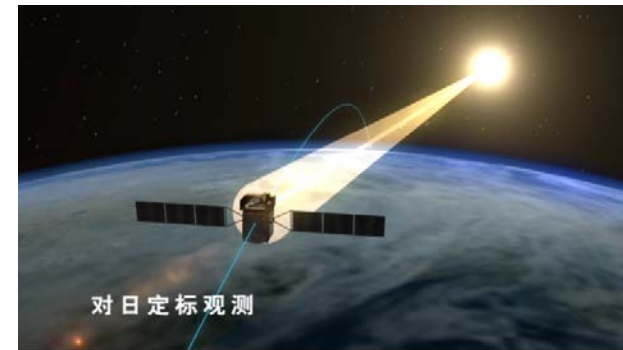
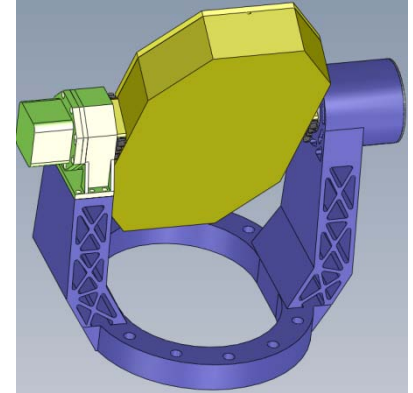
❖ CO2 Spectrometer(Once a day)

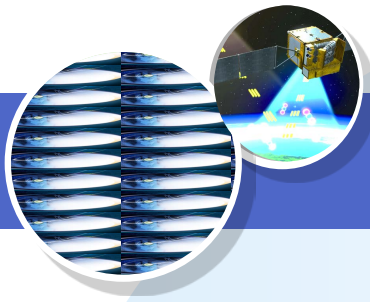
▪ Cal:

- LED in instrument for spectrometric
- Sun Calibration (by diffuser)
 - ✓ Spectrometric: look through atmosphere(limb)
 - ✓ Radiometric: look directly to

▪ Val

- TCCON





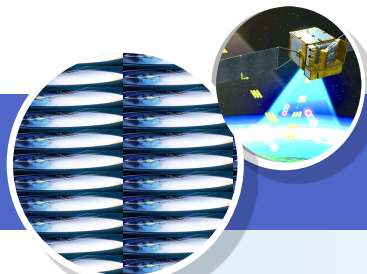
❖ CAPI(radiometric, once a month)

▪ Calibration

- LED in instrument
- Sun: for relative and absolute
- Moon for redundancy of Sun

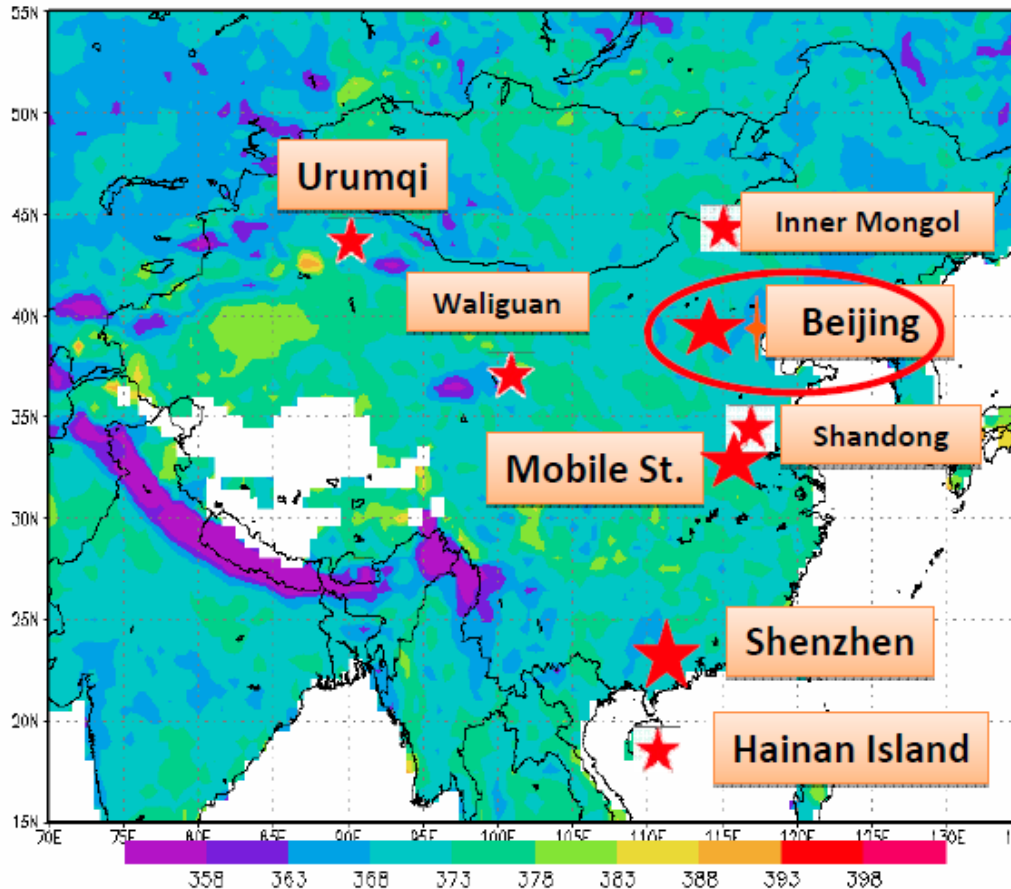
▪ TCCON for Validation





Ground Validation Stations

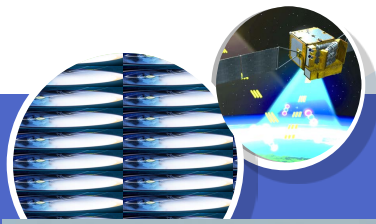
Ground-based Measurement Sites in China



Ground sites

Site	Instrument
Beijing	IFS125/HR +325mTower+7 Licor
Shenzhen	IFS125/HR CIMEL+MWR
Mobile St.	IFS 125/M
Shandong	Optical Spectrum Analyzer(OSA)
Inner Mongol	Optical Spectrum Analyzer(OSA)
Hainan island	Optical Spectrum Analyzer(OSA)
Urumqi	FGGA/LGR
Waliguan	FGGA/LGR

Calibration, Validation & a priori information



Validation Equipments



LICOR CO2 detection

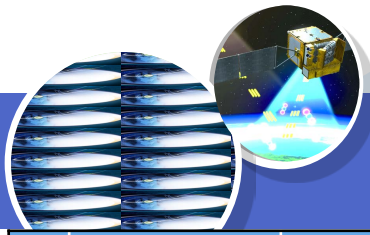


Aerosol mea



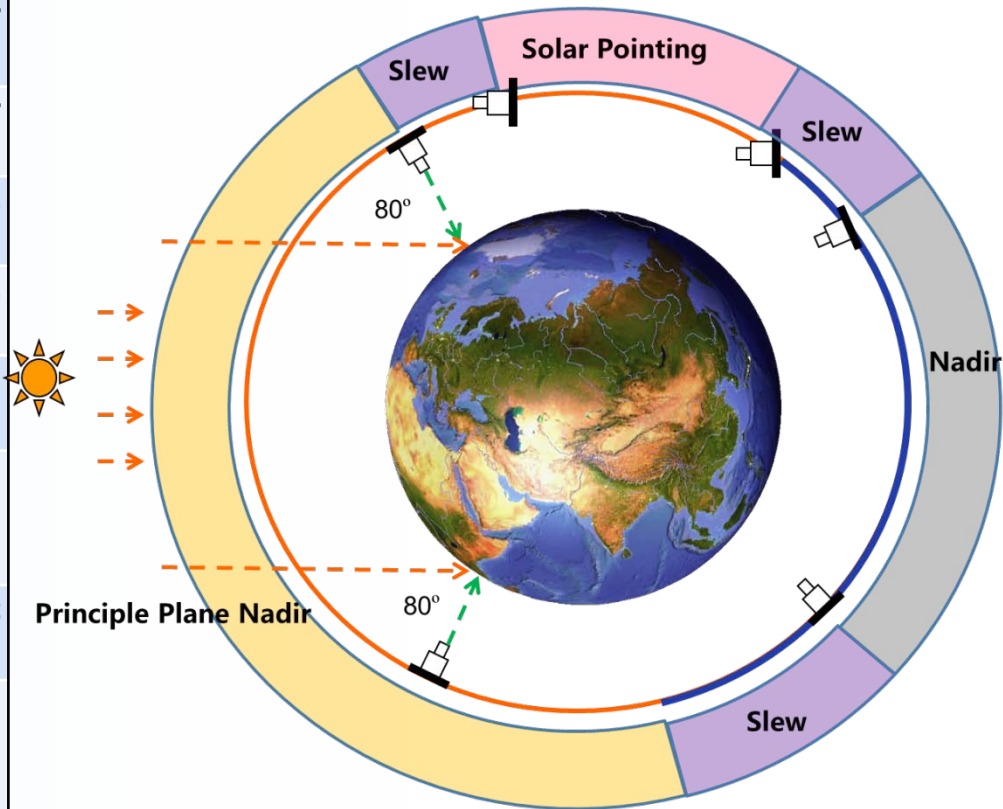
Radiation mea



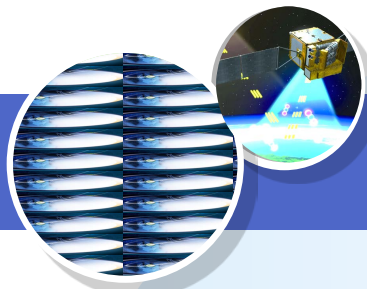


Pointing Sequence

No .	Pointing Mode	Attitude description
1	Principal Plane Nadir	Zs points along Nadir, under Principal Plane constraints
2	Sun-glint Pointing	Zs points to Sun-glint location, under Principal Plane constraints
3	Forward Nadir	+Xs points along velocity, Zs points along Nadir
4	Backward Nadir	-Xs points along velocity, Zs points along Nadir
5	Target Pointing	+Zs points to the target, small sinusoidal periodic slew on pitch axis
6	Area Steering	Fixed roll angle, maneuver on pitch axis, decreasing velocity to earth surface
7	Direct Solar Pointing	Xs points to the sun, periodic slewing, -90° mirror rotation
8	Diffusion Solar Pointing	Zs points to the sun with 15° bias on pitch axis, 180° mirror rotation
9	Moon Pointing	Zs points to the moon, small sinusoidal periodic slew on pitch axis
10	Solar Panel Pointing	-Xs points to the sun, Ys is parallel to earth equator plane
11	Attitude Slew	For pointing mode change



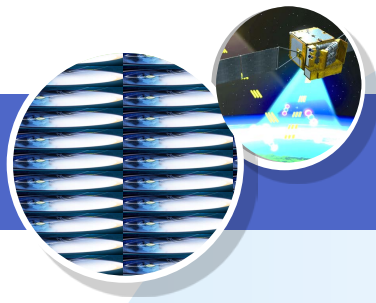
Nadir observation + Sun calibration



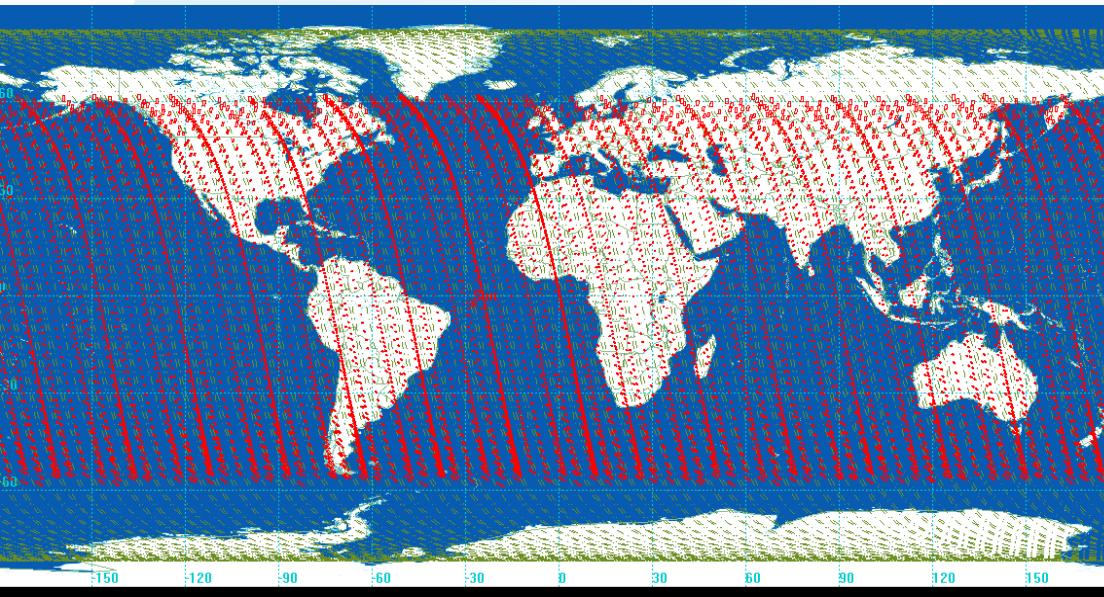
Mission Animation



Nadir Observation
High resolution observation over the land



Soundings & Coverage

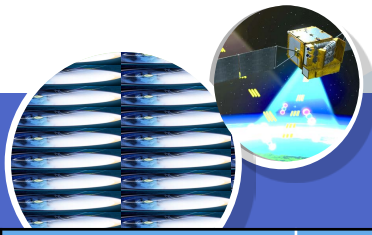


Coverage at Nadir/Sunlint mode

solar zenith angle	Nadir 80deg	Sunlint 70deg
vernal equinox	Lat [-73 79]	Lat [-66 70]
midsummer	Lat [-51 82]	Lat [-43 79]
midwinter	Lat [-82 56]	Lat [-86 46]

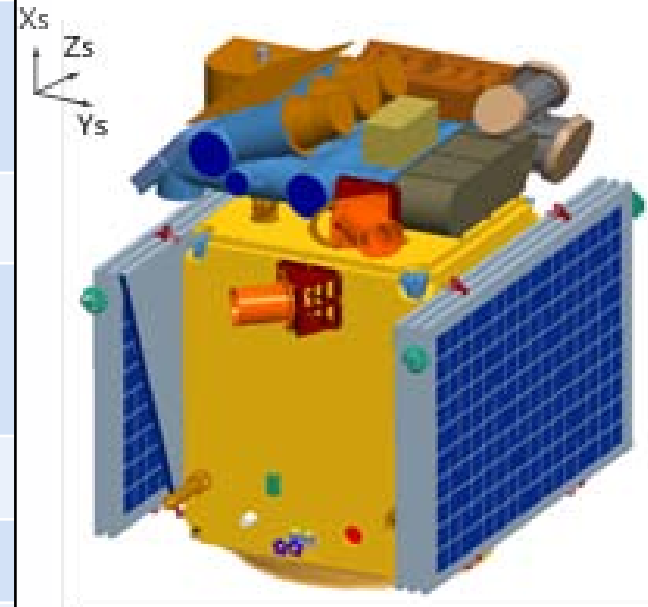
- ❖ CO2 Spectrometer
 - 3.3 Hz sample frequency
 - 20 soundings each exposure time
 - Spatial resolution: 2kmx1km
 - 2 soundings merged to increase SNR

- ❖ CAPI
 - 27 Hz sample frequency
 - 800 soundings each sample for VNIR, 500m resolution
 - 400 soundings each sample for SWIR, 1km resolution

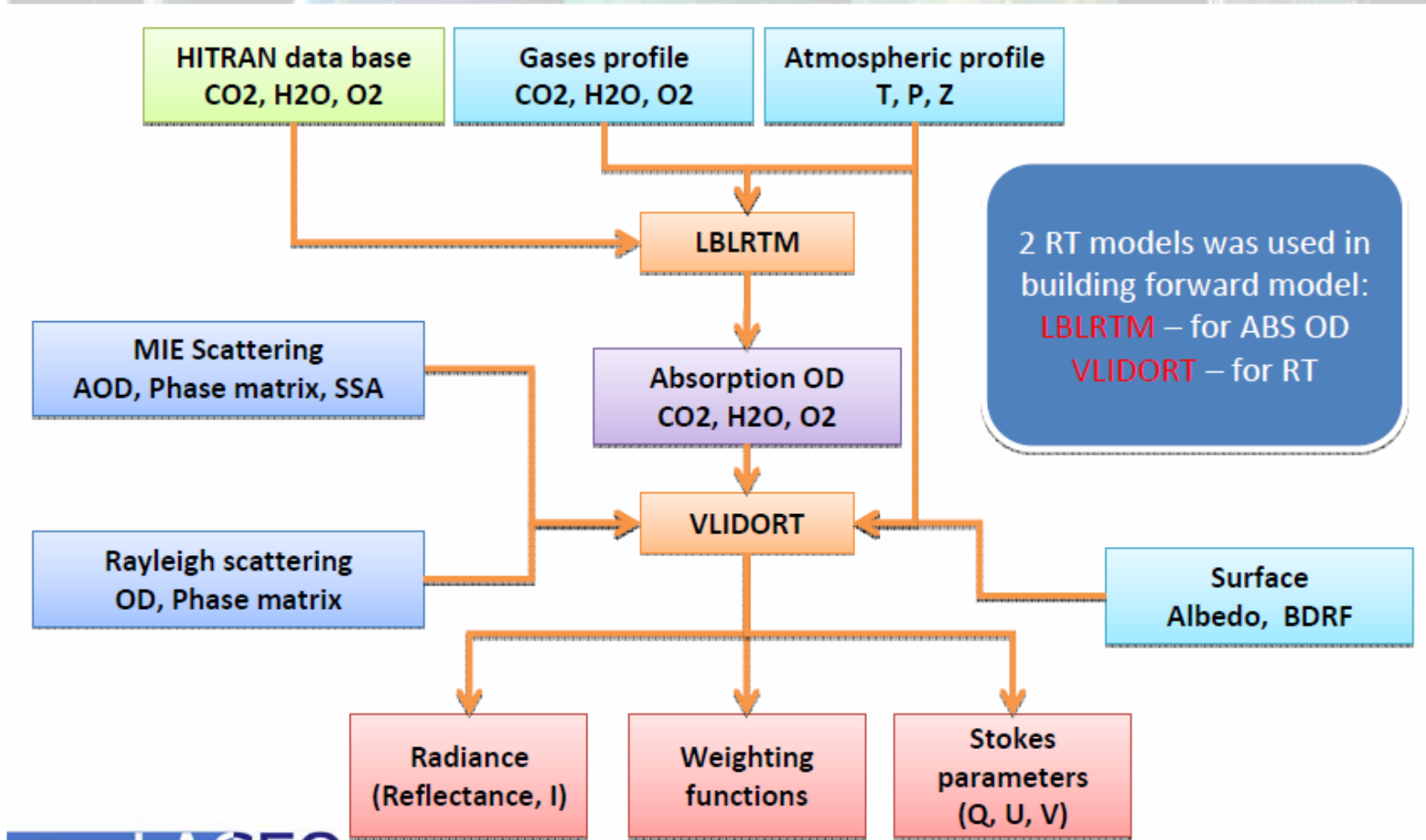


Satellite Specifications

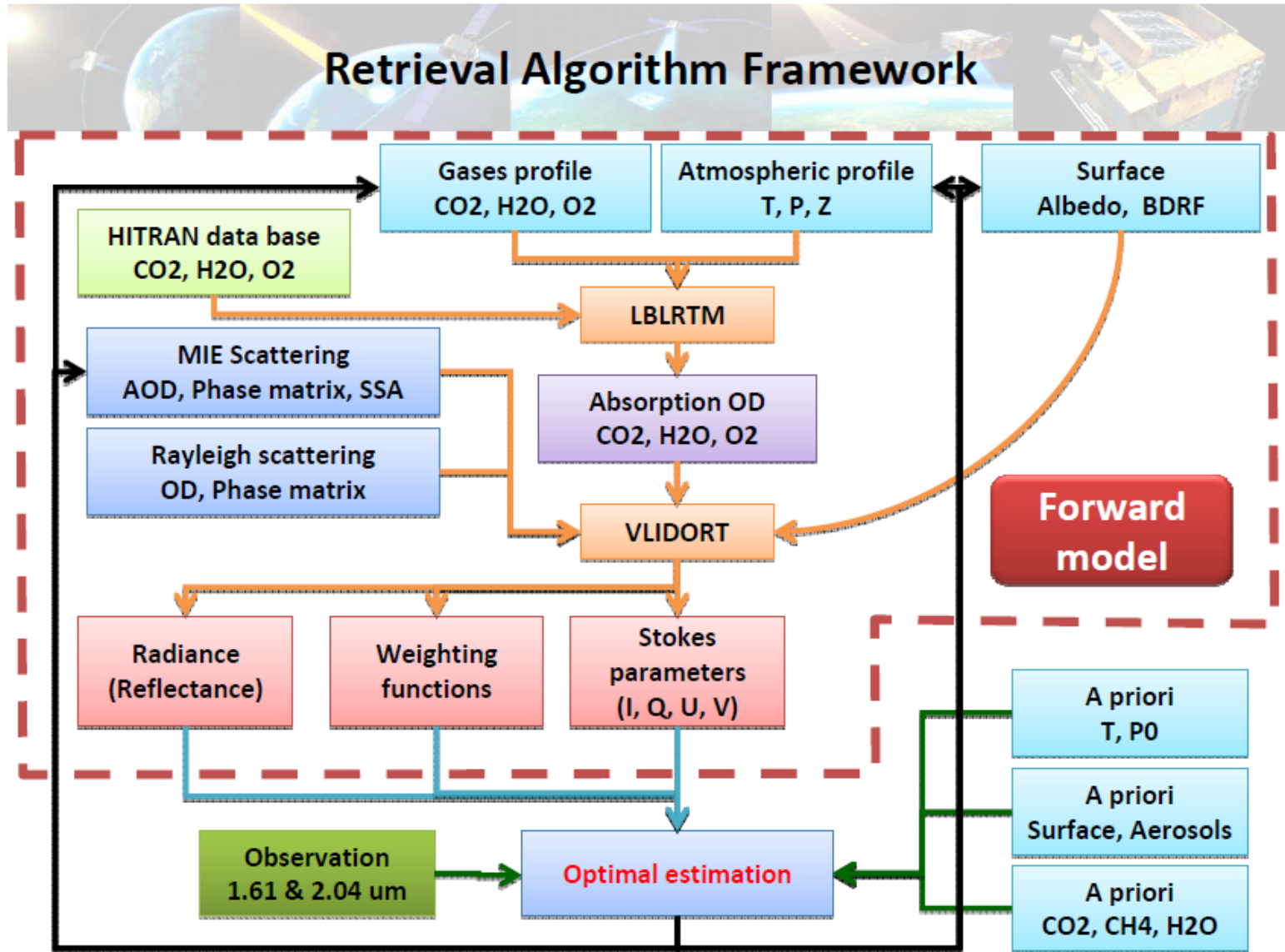
Item	Parameter
Orbit	700 km, 13:30, Local Time
Mass	~600 kg
Power	420W ~ 610W 10m ² solar array Li-battery, 80Ah
Dimensions	1500mm(Ys) × 1800mm(Zs) × 1850mm(Xs)
AOCS	Attitude pointing accuracy: ≤0.1°; Pointing stability: ≤0.001°/s; Slew rate: 180°/240s (Zs) , 150°/100s (XsYs)
Propulsion	4x1N thruster, 10kg propellant
Data Link	X-band, 64 Mbps
Data Storage	128 Gbits
TT&C	S-band, Downlink 8192bps, Uplink 2000bps
Design Life	3 years

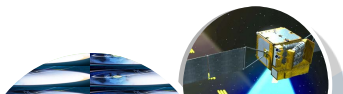


CO2 Retrieval



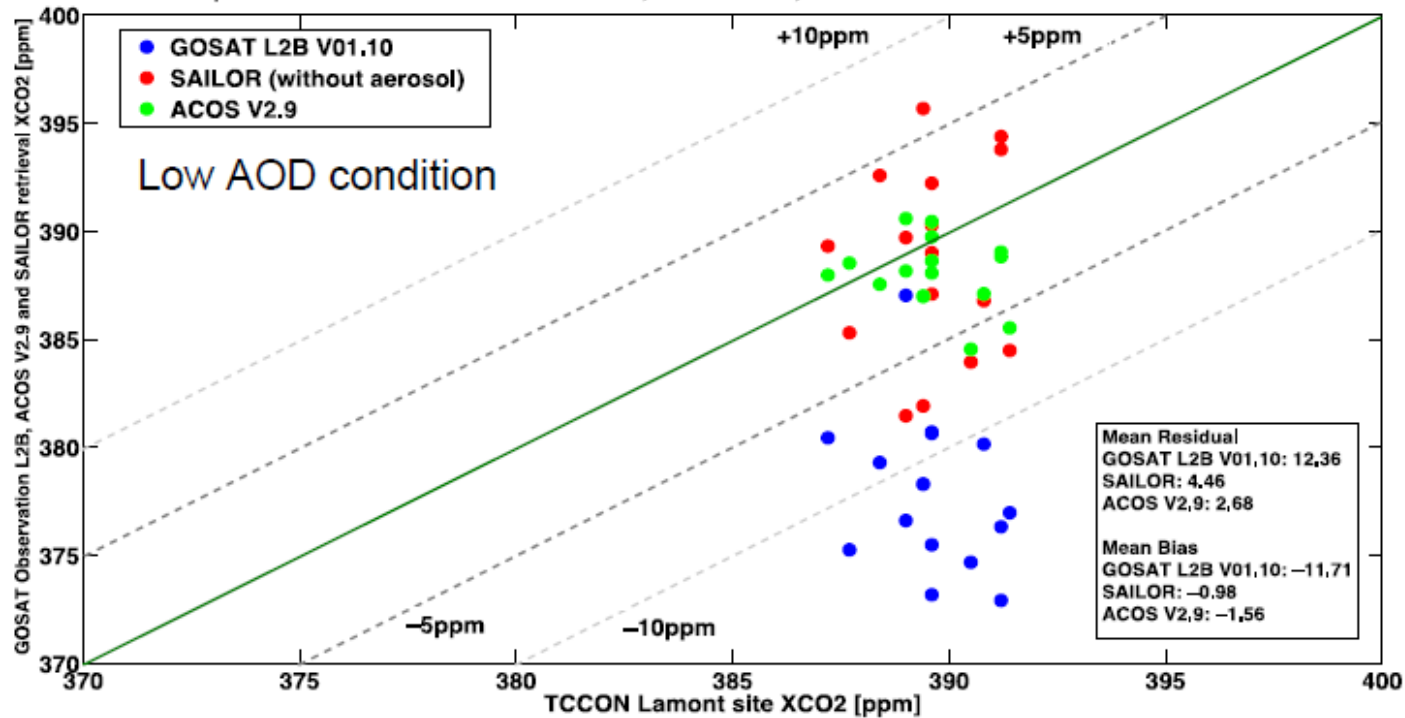
CO2 Retrieval



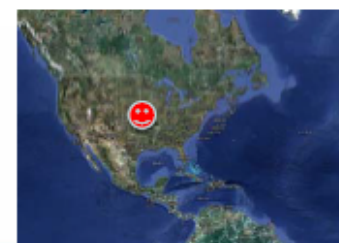


Preliminary retrieval test

The comparison between TCCON Lamont site, GOSAT L2B, ACOS V2.9 and SAILOR retrieval without aerosol



Lamont, Oklahoma, US
36.604 N, 97.486 W



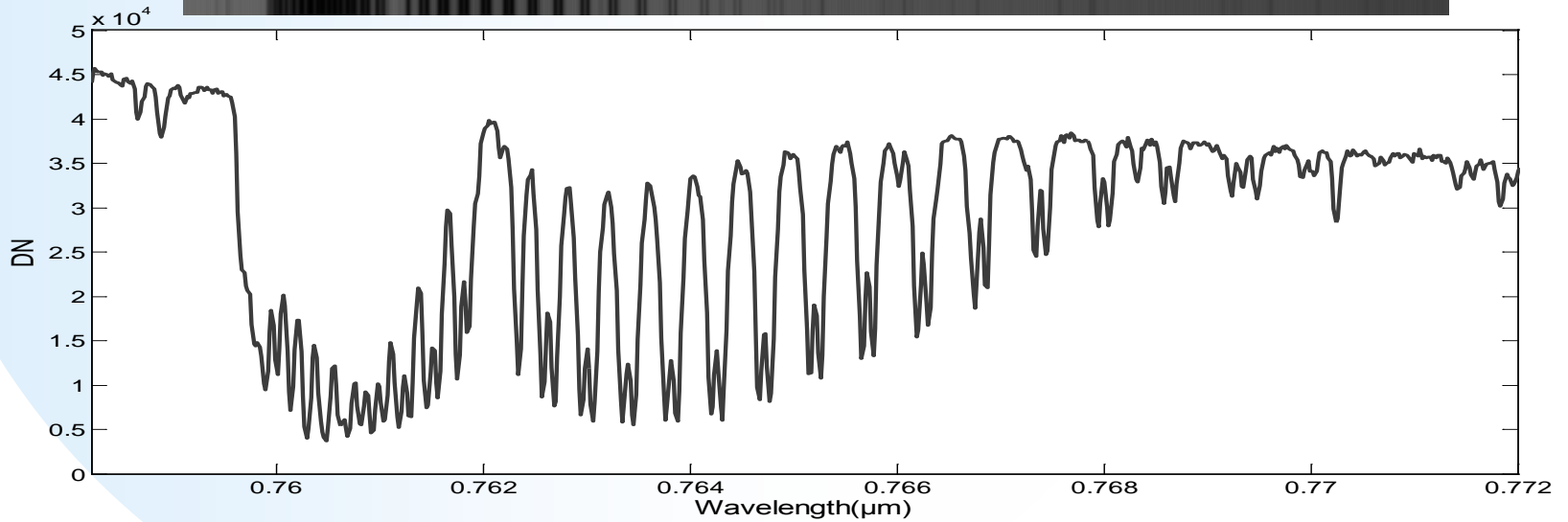
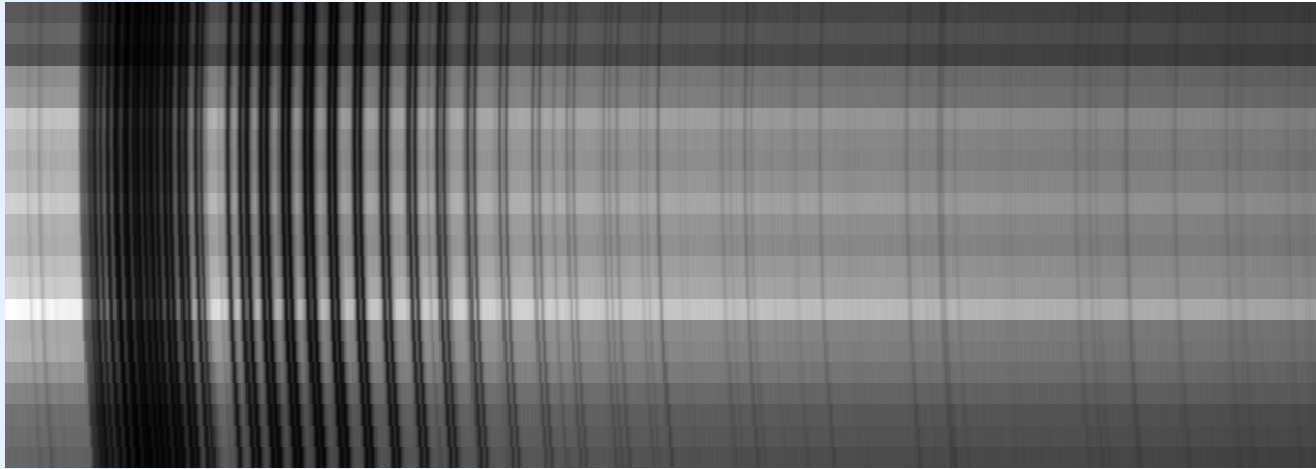
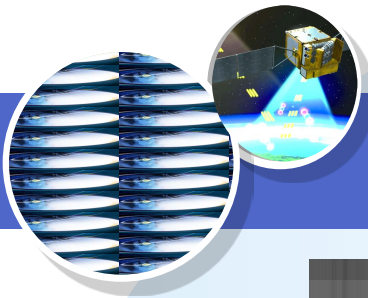
LAGEO

- Key Laboratory of the Middle Atmosphere and Global Environmental Observation

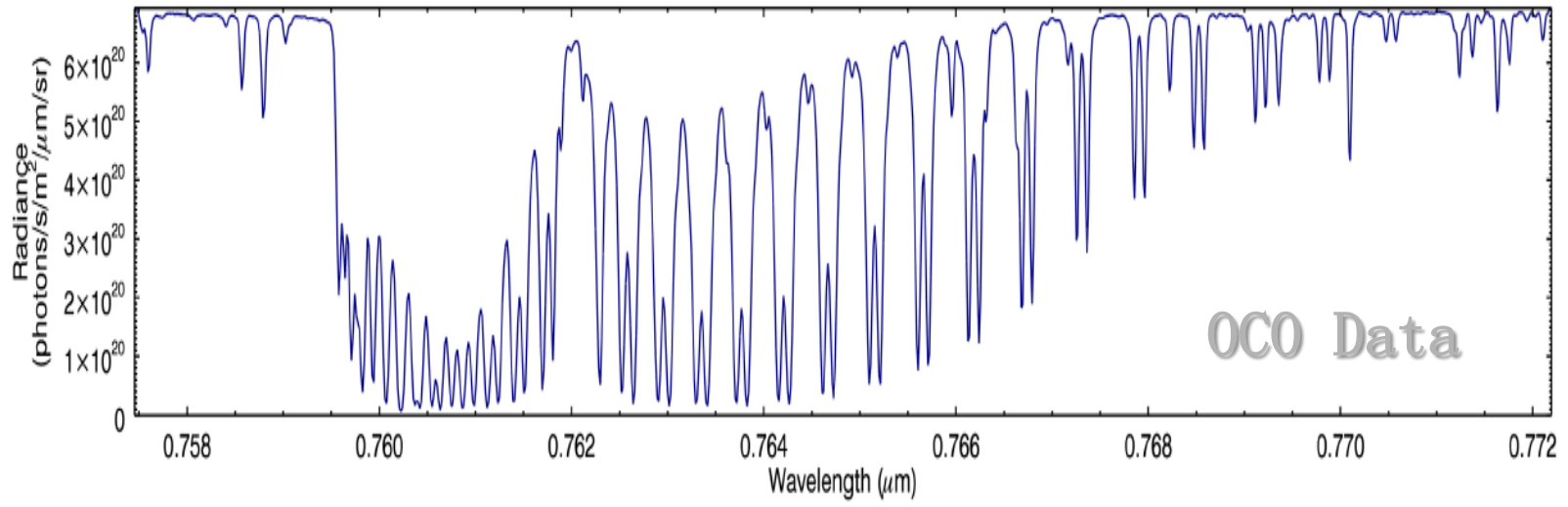
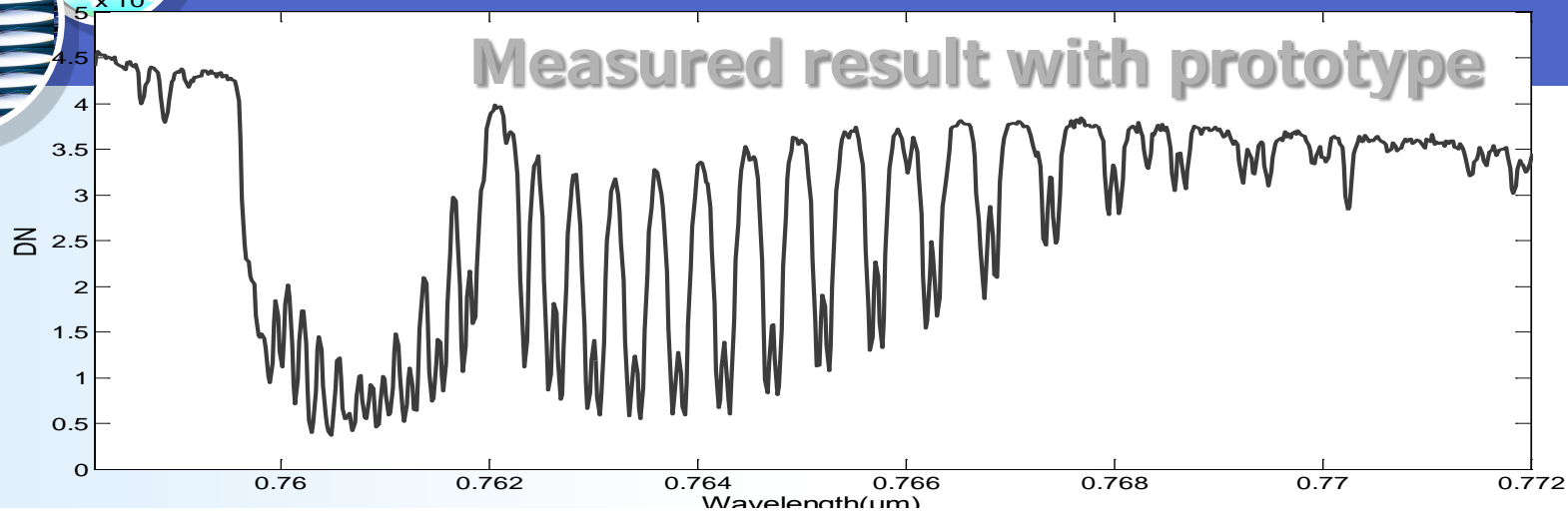
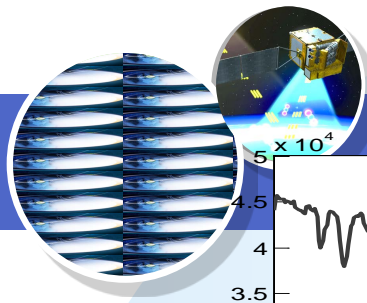
Current Status – CO2 Sounder



**760nm spectrometer prototype
with partial fore-optics and small-area grating**



Measured O₂ absorption spectrum at 760nm with prototype



Compare with OCO measured result from reference data

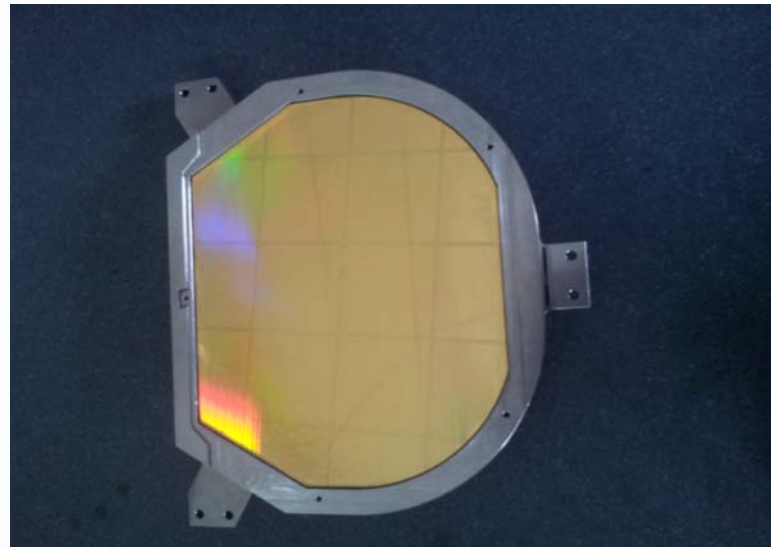
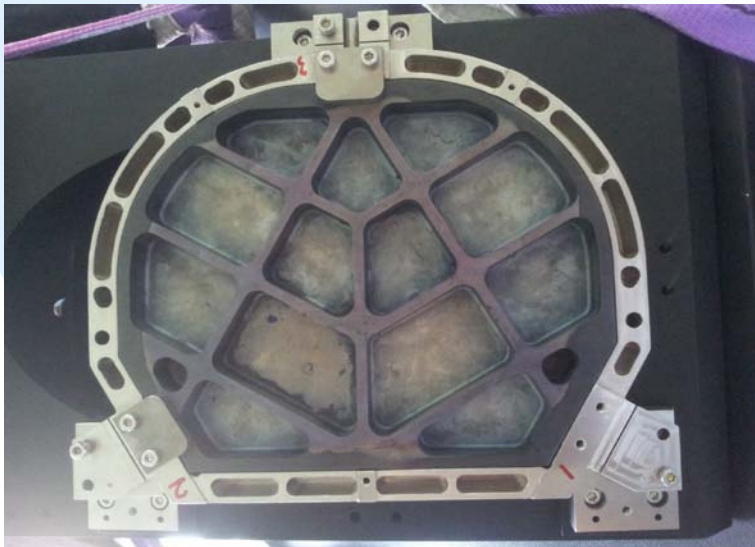
1610nm spectrometer prototype

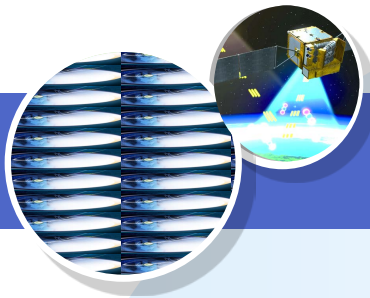
- same construction as 760nm channel
- use a large-area diffraction grating, the size is 190mm by 154mm
- use a long linear InGaAs FPA



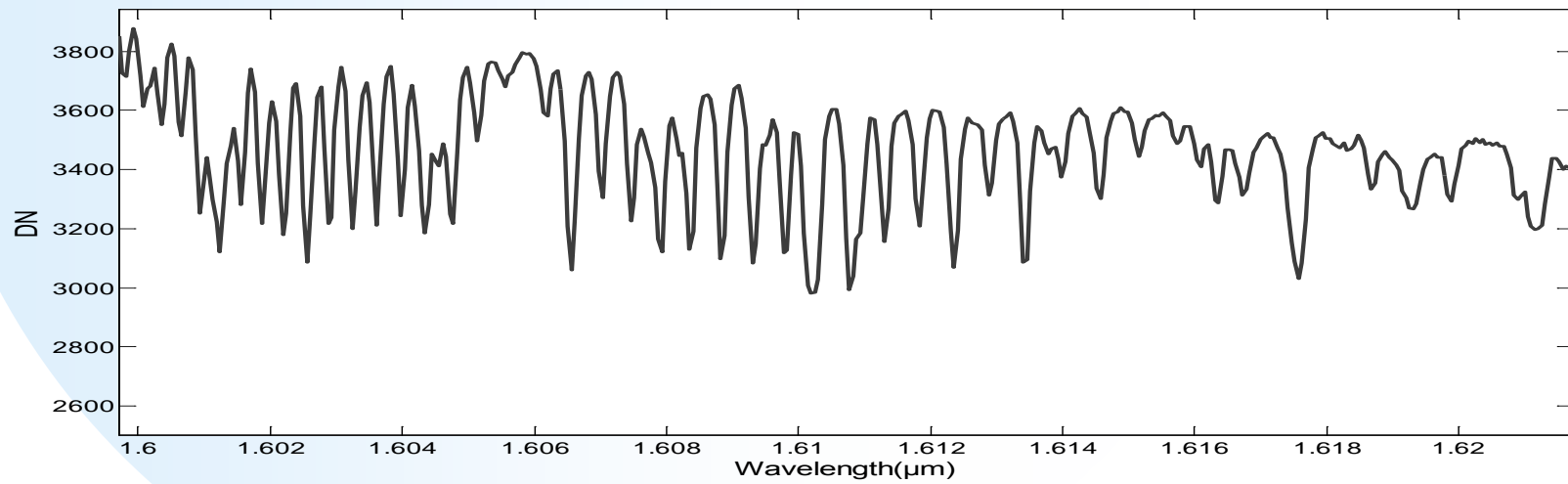
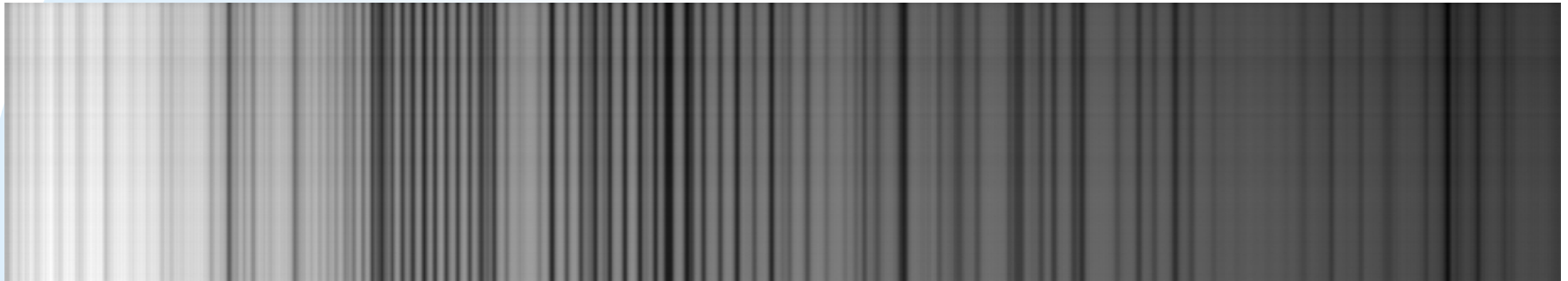
Large-area Grating Sample

- Low diffractive efficiency, high stray light and worse wavefront
- still obtain the clear spectrum
- improved technique adopted, new sample will be tested in June





Measured spectrum with 1610nm prototype



Payload Mechanical prototype

- Main frame is cast with aluminium alloy
- 960mm by 900mm by 540mm



Satellite Tests

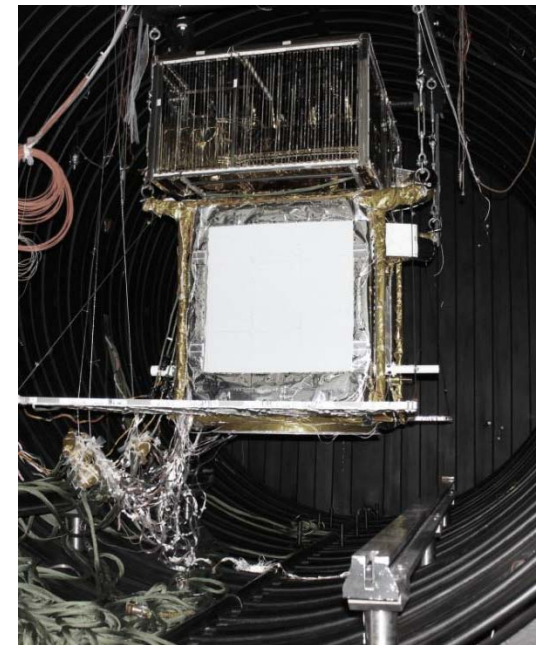
- ❖ Satellite electrical interface tests in 2012.8
- ❖ Satellite mechanical test in 2013.1
- ❖ Satellite thermal test in 2013.4



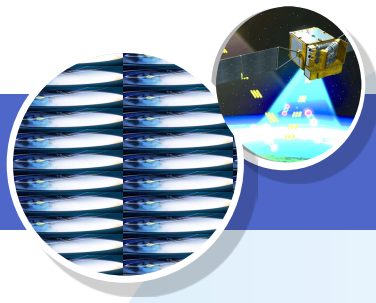
EM of Onboard Computer



Satellite on vibration bench



Satellite in KM3



Development Plan

2011.2	Kick off of project
2011.9	SRR
2012.9	PDR
2013.6	Kick off phase C
2014.5	CDR
2015.6	SRR

TanSat is to be launched in 2015!

The background features a stylized world map in shades of blue and white. Overlaid on the map are several glowing, curved lines in yellow and orange, suggesting global connectivity or data flow. On the left side, there is a grid of small squares in various shades of blue and purple. The overall aesthetic is modern and technological.

Thank You !